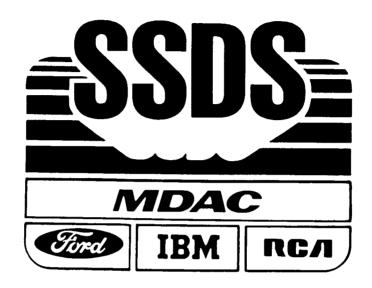
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SPACE STATION DATA SYSTEM ANALYSIS/ARCHITECTURE STUDY

Task 5 - Program Plan

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SPACE STATION DATA SYSTEM ANALYSIS/ARCHITECTURE STUDY

Task 5 - Program Plan

DECEMBER 1985

MDC H1343A REPLACES MDC H2022 DATED OCTOBER 1985

MCDONNELL DOUGLAS ASTRONAUTICS COMPANY-HUNTINGTON BEACH

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PREFACE

The McDonnell Douglas Astronautics Company has been engaged in a Space Station Data System Analysis/Architecture Study for the National Aeronautics and Space Administration, Goddard Space Flight Center. This study, which emphasized a system engineering design for a complete, end-to-end data system, was divided into six tasks:

- Task 1. Functional Requirements Definition
- Task 2. Options Development
- Task 3. Trade Studies
- Task 4. System Definitions
- Task 5. Program Plan
- Task 6. Study Maintenance

McDonnell Douglas was assisted by the Ford Aerospace and Communications Corporation, IBM Federal Systems Division and RCA in these Tasks. The Task inter-relationship and documentation flow are shown in Figure 1.

This report was prepared for the National Aeronautics and Space Administration Goddard Space Flight Center under Contract No. NASS-28082

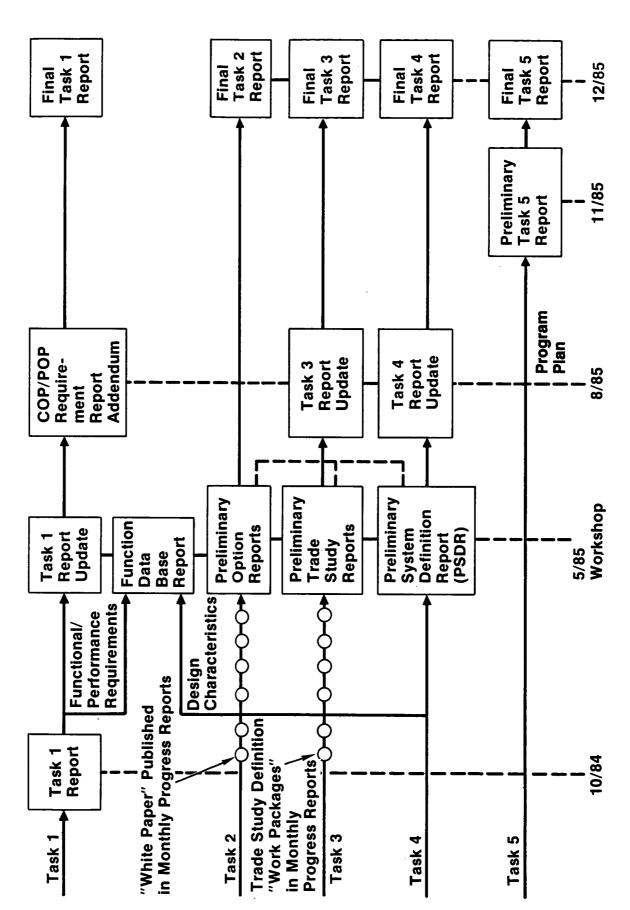
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SSDS A/A DOCUMENTATION SCHEDULE

VHG598



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GLOSSARY

Α Automatic A&R Automation and Robotics A/A Analysis/Architecture A/D Advanced Development A/L Airlock A/N Alphanumeric AC&S Attitude Control System ACA Attitude Control Assembly ACO Administrative Contracting Officer ACS Attitude Control and Stabilization Attitude Control System/Communications ACS/COM **ACTS** Advanced Communications Technology Satellite AD Ancillary Data AD Advanced Development ADOP Advanced Distributed Onboard Processor ADP Advanced Development Plan **AFOSR** Air Force Office of Scientific Research AFP Advanced Flexible Processor AFRPL Air Force Rocket Propulsion Laboratory AGC Automatic Gain Control AGE Attempt to Generalize ΑI Artificial Intelligence AIE Ada Integrated Environment AIPS Advanced Information Processing System ALT Air Lock One ALS Alternate Landing Site ALS/N Ada Language System/Navy AMIC Automated Management Information Center ANSI American National Standards Institute AOS Acquisition of Signal AP Automatic Programming APD Avalanche Photo Diode

Ada Programming Support Environment

Ames Research Center



APSE

ARC

ART Automated Reasoning Tool

ASCII American Standard Code for Information Exchange

ASE Airborne Support Equipment

ASTROS Advanced Star/Target Reference Optical Sensor

ATAC Advanced Technology Advisory Committee

ATC Air Traffic Control
ATP Authority to Proceed

ATPS Advanced Telemetry Processing System

ATS Assembly Truss and Structure

AVMI Automated Visual Maintenance Information

AWSI Adoptive Wafer Scale Integration

B Bridge

BARC Block Adaptive Rate Controlled

BB Breadboard

BER Bit Error Rate

BIT Built-in Test

BITE Built-in Test Equipment

BIU Buffer Interface Unit

BIV Bus Interface Unit

BIU Built-in Unit

BMD Ballistic Missile Defense

BTU British Thermal Unit

BW Bandwidth C Constrained

C Command and Control

C Command, Control, and Communication

Command, Control, Communication, and Intelligence

C&DH Communications and Data Handling

C&T Communication and Tracking Subsystem

C&T Communications and Tracking

C&W Control and Warning

C/L Checklist

CA Customer Accommodation

CAD Computer-Aided Design

CAE Computer-Aided Engineering
CAIS Common APSE Interface Set

CAM Computer-Aided Manufacturing

CAMAC Computer Automatic Measurement and Control

CAP Crew Activities Plan

CASB Cost Accounting Standard Board

CASE Common Application Service Elements

CATL Controlled Aceptance Test Library

CBD Commerce Business Daily

CBEMA Computer and Business Equipment Manufacturing Association

CCA Cluster Coding Algorithm

CCB Contractor Control Board

CCB Configuration Control Board

CCC Change and Configuration Control

CCD Charge-Coupled Device

CCITT Consultive Committee for International Telegraph and Telephone

CCITT Coordinating Committee for International Telephony and Telegraphy

CCMS Checkout Control and Monitor System

CCR Configuration Change Request

CCSDS Consultative Committee for Space Data System

CCTV Closed-Circuit Television cd/M² Candelas per square Meter

CDG Concept Development Group

CDMA Code Division Multiple Access
CDOS Customer Data Operations System

CDR Critical Design Review

CDS Control Data Subsystem

CE Conducted Emission

CEI Contract End-Item

CER Cost Estimating Relationship
CFR Code of Federal Regulations

CFS Cambridge File Server

CG Center of Gravity

CIE Customer Interface Element

CIL Critical Item List

CIU Customer Interface Unit CLAN Core Local Area Network

CM Configuration Management

CM Center of Mass

CMDB Configuration Management Data Base

CMG Control Moment Gyro

CMOS Complementary Metal-Oxide Semiconductor

CMS Customer Mission Specialist
CMU Carnegie-Mellon University

CO Contracting Officer

COF Component Origination Form

COL Controlled Operations Library

COMM Commercial Missions
COP Co-orbital Platform

COPCC Coorbit Platform Control Center
COPOCC COP Operations Control Center

COTS Commercial Off-the-Shelf Software
CPCI Computer Program Configuration Item

CPU Central Processing Unit

CQL Channel Queue Limit
CR Compression Ratio

CR Change Request

CR&D Contract Research and Development

CRC Cyclic Redundancy Checks

CRF Change Request Form

CRSS Customer Requirements for Standard Services

CRT Cathode Ray Tube

CS Conducted Susceptibility

CSD Contract Start Date

CSDL Charles Stark Draper Laboratory

CSMA/CD/TS Carrier-Sense Multiple with Access/Collision Detection and Time

Slots

CSTL Controlled System Test Library
CTA Computer Technology Associates
CTE Coefficient of Thermal Expansion

CUI Common Usage Item

CVSD Code Variable Slope Delta (Modulation)

CWG Commonality Working Group

D&B Docking and Berthing

DADS Digital Audio Distribution System
DAIS Digital Avionics Integration System

DAR Defense Acquisition Regulation

DARPA Defense Advanced Research Projects Agency

DB Data Base

DBA Data Base Administrator

DBML Data Base Manipulation Language

DBMS Data Base Management System

DCAS Defense Contract Administrative Services

DCDS Distributed Computer Design System

DCR Data Change Request

DDBM Distributed Data Base Management

DDC Discipline Data Center

DDT&E Design, Development, Testing, and Engineering

DEC Digital Equipment Corp.

DES Data Encryption Standard

DFD Data Flow Diagram

DGE Display Generation Equipment

DHC Data Handling Center
DID Data Item Description
DIF Data Interchange Format

DMA Direct Memory Access

DMS Data Management System

DoD Department of Defense

DOMSAT Domestic Communications Satellite System

DOS Distributed Operating System
DOT Department of Transportation

DPCM Differential Pulse Code Modulation

DPS Data Processing System

DR Discrepancy Report
DR Data Requirement

DRAM Dynamic Random-Access Memory
DRD Design Requirement Document

DS&T Development Simulation and Training

DSDB Distributed System Data Base

DSDL Data Storage Description Language

DSDS Data System Dynamic Simulation

DSIT Development, Simulation, Integration and Training

DSN Deep-Space Network

DTC Design to Cost

DTC/LCC Design to Cost/Life Cycle Cost

DTG Design To Grow

E/R Entity/Relationship

EADI Electronic Attitude Direction Indicator

ECC Error Correction Codes

ECLSS Environmental Control and Life-Support System

ECMA European Computers Manufacturing Assoc.

ECP Engineering Change Proposals

ECS Environmental Control System

EDF Engineering Data Function

EEE Electrical, Electronic, and Electromechanical

EHF Extremely High Frequency

EHSI Electronic Horizontal Situation Indicator

EIA Electronic Industry Association

EL Electroluminescent

EM Electromagnetic

EMC Electromagnetic Compatibility

EMCFA Electromagnetic Compatibility Frequency Analysis

EME Earth Mean Equator

EMI Electromagnetic Interference

EMR Executive Management Review

EMS Engineering Master Schedule

EMU Extravehicular Mobility Unit

EMUDS Extravehicular Maneuvering Unit Decontamination System

E0 Electro-optic

EOL End of Life

EOS Earth Observing System

EPA Environmental Protection Agency

EPS Electrical Power System

ERBE Earth Radiation Budget Experiment

ERRP Equipment Replacement and Refurbishing Plan

ESR Engineering Support Request

ESTL Electronic Systems Test Laboratory

EVA Extravehicular Activity

F/T Fault Tolerant

FACC Ford Aerospace and Communications Corporation

FADS Functionally Automated Database System

FAR Federal Acquisition Regulation FCA Functional Configuration Audit

FCOS Flight Computer Operating System

FCR Flight Control Rooms

FDDI Fiber Distributed Data Interface

FDF Flight Dynamics Facility

FDMA Frequency-Division Multiple Access

FEID Flight Equipment Interface Device

FETMOS Floating Gate Election Tunneling Metal Oxide Semiconductor

FF Free Flier

FFT Fast Fourier Transform

FIFO First in First Out

FIPS Federal Information Processing Standards

fl foot lambert - Unit of Illumination

FM Facility Management

FMEA Failure Modes and Effects Analysis

FMECA Failure Mode Effects and Criticality Analysis

FO Fiber-Optics

FO/FS/R Fail-Operational/Fail Safe/Restorable

FOC Fiber-Optic Cable

FOOB Fiber-Optic Data Bus

FODS Fiber Optic Demonstration System

FPR Federal Procurement Regulation

FOR Formal Qualification Review

FSD Full-Scale Development
FSE Flight Support Equipment

FSED Full Scale Engineering Development

FSIM Functional Simulator

FSW Flight Software

FTA Fault Tree Analysis

FTMP Fault Tolerant Multi-Processor

FTSC Fault Tolerant Space Computer

GaAs Gallium Arsenide

GaAsP Gallium Arsenic Phosphorus
GaInP Gallium Indium Phosphorus

GaP Gallium Phosphorous

GAPP Geometric Arithmetic Parallel Processor

Gbps Gigabits Per Second

GBSS Ground Based Support System

GEO Geosynchronous Earth Orbit

GEP Gas Election Phosphor

GFC Ground Forward Commands

GFE Government-Furnished Equipment

GFP Government-Furnished Property

GFY Government Fiscal Year

GIDEP Government/Industry Data Exchange Program

GMM Geometric Math Model

GMS Geostationary Meteorological Satellite

GMT Greenwich Mean Time

GMW Generic Maintenance Work Station
GN&C Guidance. Navigation. and Control

GPC General-Purpose Computer
GPP General-Purpose Processor

GPS Global Positioning System

GRO Gamma Ray Observatory
GSC Ground Service Center

GSE ground Support Equipment

GSFC (Robert H.) Goddard Space Flight Center

GTOSS Generalized Tethered Object System Simulation

H/W Hardware

HAL High-Order Algorithmic Language
HDDR Help Desk Discrepancy Report
HDDR High Density Digital Recording
HEP Heterogeneous Element Processor

HFE Human Factors Engineering

HIPO Hierarchical Input Process Output
HIRIS High Resolution Imaging Spectrometer

HM1 Habitation Module One

HM Habitation Module

HOL High Order Language
HOS High Order Systems

HPP High Performance Processors

HRIS High Resolution Imaging Spectrometer

I Interactive

I/F Interface

I/O Input/Output

IC Intercomputer

ICAM Integrated Computer-Aided Manufacturing

ICB Internal Contractor Board
ICD Interface Control Document

ICOT Institute (for new generation) Computer Technology

ICS Interpretive Computer Simulation

ID Interface Diagram

ID Identification

IDM Intelligent Database Machine

IDMS Information and Data Management System

IEEE Institute of Electrical and Electronic Engineers

IEMU Integrated Extravehicular Mobility Unit

IF Intermediate frequency

IFIPS International Federation of Industrial Processes Society

ILD Injector Laser Diode

IMU Inertial Measurement Unit

INS Inertial Navigation System

IOC Initial Operating Capability

IOP Input/Output Processor

IPCF Interprocess Communications Facility

IPC Interprocesses Communication

IPL Initial Program Load

IPR Internal Problem Report

IPS Instrument Pointing System

IR Infrared

IR&D Independent Research and Development

IRN Interface Revision Notices

ISA Inertial Sensor Assembly

ISA Instruction Set Architecture

ISDN Integration Services Digital Network

ISO International Standards Organization

ITAC-O Integration Trades and Analysis-Cycle O

ITT International Telegraph and Telephone

IV&V Independent Validation and Verification

IVA Intravehicular Activity
IWS Intelligent Work Station

JPL Jet Propulsion Laboratory

JSC (Lyndon B.) Johnson Space Center

KAPSE Kernal APSE

KEE Knowledge Engineering Environment

KIPS Knowledge Information Processing System

KOPS Thousands of Operations Per Second

KSA Ku-band, Single Access

KSC (John F.) Kennedy Space Center

Kbps Kilobits per second

Kipc Thousand instructions per cycle

LAN Local-Area Network

LaRC Langley Research Center

LCC Life-Cycle Cost

LCD Liquid Crystal Display

LDEF Long-Duration Exposure Facility

LDR Large Deployable Reflector

LED Light-Emitting Diode

LEO Low Earth Orbit

LeRC Lewis Research Center

LIDAR Laser-Instrument Distance and Range

LIFO Last In First Out

LIPS Logical Inferences Per Second

LISP List Processor LisP List Processor

LLC Logical Link Control

LMI LISP Machine Inc.
LN₂ Liquid Nitrogen

LNA Low-noise Amplifier

LOE Level of Effort

LOE Low-earth Orbit Environments

LOS Loss of Signal

LPC Linear Predictive Coding
LPS Launch Processing System

LRU Line-Replaceable unit

LSA Logistic Support Analysis

LSAR Logistic Support Analysis Report

LSE Language Sensity Editors
LSI Large-scale Integration

LTV Aerospace and Defense Company, Vought Missiles Advanced

Programs Division

LZPF Level O Processing Facility

M Manual

μP Microprocessor MA Multiple Access

MA Managing Activity

MAPSE Minimum APSE

Mbps Million Bits Per Second
MBPS Million Bits Per Second
MCAIR McDonnell Aircraft Company

MCC Mission Control Center

MCC Microelectronics and Computer Technology Corp.

MCDS Management Communications and Data System

MCM Military Computer Modules

MCNIU Multi-compatible Network Interface Unit

MDAC-HB McDonnell Douglas Astronautics Company-Huntington Beach

MDAC-STL McDonnell Douglas Astronautics Company-St. Louis

MDB Master Data Base

MDC McDonnell Douglas Corporation

MDMC McDonnell Douglas Microelectronics Center

MDRL McDonnell Douglas Research Laboratory

MFLOP Million Floating Point Operations

MHz Million Hertz

MIMO Multiple-Input Multiple-Output

MIPS Million (machine) Instructions Per Second

MIT Massachusetts Institute of Technology

MITT Ministry of International Trade and Industry

MLA Multispectral Linear Array

MMI Man Machine Interface

MMPF Microgravity and Materials Process Facility

MMS Module Management System
MMS Momentum Management System

MMU Mass Memory Unit

MMU Manned Maneuvering Unit

MNOS Metal-Nitride Oxide Semiconductor

MOC Mission Operations Center

MOI Moment of Inertia

MOL Manned Orbiting Laboratory
MOS Metal Oxide Semiconductor

MPAC Multipurpose Application Console

MPS Materials, Processing in Space

MPSR Multi-purpose Support Rooms

MRMS Mobile Remote Manipulator System
MRWG Mission Requirements Working Group

MSFC (George C.) Marshall Space Flight Center

MSI Medium-Scale Integration

MSS Multispectral Scanner

MTA Man-Tended Approach

MTBF Mean Time Between Failures

MTTR Mean Time to Repair
MTU Master Timing Unit

NASA National Aeronautics and Space Administration

NASCOM NASA Communications Network
NASPR NASA Procurement Regulation

NBO NASA Baseline

NBS National Bureau of Standards

NCC Network Control Center

NFSD NASA FAR Supplement Directive

NGT NASA Ground Terminals

NHB NASA Handbook

NISDN NASA Integrated System Data Network

NIU Network Interface Unit

NL National Language

NLPQ National Language for Queuing Simulation

NMI NASA Management Instruction

NMOS N-Channel Metal-Oxide Semiconductor

NMR N-Modular Redundant

NOS Network Operating System

NS Nassi-Schneidermann

NSA National Security Administration

NSF National Science Foundation

NSTS National Space Transportation System

NTDS Navy Tactical Data System

NTE Not To Exceed

NTRL NASA Technology Readiness Level

NTSC National Television Standards Committee

Nd: YAG Neodynium Yttrium Aluminum Garnet (laser type)

O&M Operations and Maintenance

0/B Onboard

OASCB Orbiter Avionics Software Control Board

OCN Operations and Control Network, Operational Control Networks

ODB Operational Data Base

ODBMS Onboard Data Base Management System

OEL Operating Events List

OES Operating Events Schedule

OID Operations Instrumentation Data
OLTP On Line Transaction Processing

OMCC Operations Management and Control Center

OMV Orbital Maneuvering Vehicle

ONR Office of Naval Research
ORU Orbital Replacement Unit

OS Operating System

OSE Orbit Support Equipment
OSI Open Systems Interconnect

OSM Orbital Service Module

OSSA Office of Space Science and Applications
OSTA Office of Space and Terrestrial Application
OSTDS Office of Space Tracking and Data Systems

OTV Orbital Transfer Vehicle

P&SA Payload and Servicing Accommodations

P/L Payload

PA Product Assurance

PAM Payload Assist Module

PASS Primary Avionics Shuttle Software

PBX Private Branch Exchange

PC Personal Computer

PCA Physical Configuration Audit

PCA Program Change Authorization

PCM Pulse Code Modulation
PCR Program Change Request

PDP Plazma Display Panel

PDR Preliminary Design Review

PDRD Program Definition and Requirements Document

PDRSS Payload Deployment and Retrieval System Simulation

PILS Payload Integration Library System

PIN Personal Identification Number

PLAN Programmable Logic Array
PLAN Payload Local Area Network
PLSS Payload Support Structure

PMAD Power Management and Distribution
PMC Permanently Manned Configuration

PN Pseudonoise

POCC Payload Operations Control Center

POP Polar Orbiter Platform

POPCC Polar Orbit Platform Control Center

POPOCC POP Operations Control Center
PRISM Prototype Inference System
PSA Problem Statement Analyzer
PSA Preliminary Safety Analysis

PSCN Program Support Communications Network

PSL Problem Statement Language

PTR Problem Trouble Report

QA Quality Assurance

R Restricted

R&D Research and Development

R&QA Reliability and Quality Assurance

R/M/A Reliability/Maintainability/Availability

R/T Real Time

RAD Unit of Radiation

RAM Random Access Memory

RAP Relational Associative Processor

RC Ring Concentrator
RCA RCA Corporation

RCS Reaction Control System

RDB relational Data Base

RDC Regional Data Center

REM Roentgen Equivalent (man)

RF Radio Frequency

RFC Regenerative Fuel Cell

RFI Radio Frequency Interference

RFP Request for Proposal

RGB Red-Green-Blue

RID Review Item Disposition

RID Revision Item Description

RISC Reduced Instruction Set Computer

RMS Remote Manipulator System

RMSE Root Mean Square Error

RNET Reconfiguration Network

ROM Read Only Memory

ROTV Reuseable Orbit Transfer Vehicle

RPMS Resource Planning and Management System

RS Reed-Solomon

RSA Rivest, Skamir and Adleman (encryption method)

RTX Real Time Execution

S&E Sensor and Effector

S/C Spacecraft

S/W Software

SA Single Access

SA Structured Analysis

SAAX Science and Technology Mission

SAE Society of Automotive Engineers

SAIL Shuttle Avionics Integration Laboratory

SAIS Science and Applications Information System

SAR Synthetic Aperture Radar
SAS Software Approval Sheet

SASE Specific Application Service Elements

SATS Station Accommodations Test Set

SBC Single Board Computer

SC Simulation Center

SCR Software Change Request

SCR Solar Cosmic Ray

SCS Standard Customer Services

SDC Systems Development Corporation

SDP Subsystem Data Processor

SDR System Design Review

SDTN Space and Data Tracking Network

SE&I Systems Engineering and Integration

SEI Software Engineering Institute

SESAC Space and Earth Scientific Advisory Committee

SESR Sustaining Engineering System Improvement Request

SESS Software Engineering Standard Subcommittee

SEU Single Event Upset

SFDU Standard Format Data Unit

SI International System of Units

SIB Simulation Interface Buffer

SIFT Software Implemented Fault Tolerance

SIMP Single Instruction Multi-Processor

SIRTF Shuttle Infrared Telescope Facility

SLOC Source Lines of Code

SMC Standards Management Committee

SMT Station Management

SNA System Network Architecture

SNOS Silicon Nitride Oxide Semiconductor

SNR Signal to Noise Ratio

SOA State Of Art

SOPC Shuttle Operations and Planning Complex

SOS Silicon On Saphire

SOW Statement of Work

SPC Stored Payload Commands

SPF Software Production Facility

SPF Single-Point Failure

SPR Spacelab Problem Reports
SPR Software Problem Report

SQA Software Quality Assurance

SQAM Software Quality Assessment and Measurement

SQL/DS SEQUEL Data System

SRA Support Requirements Analysis
SRAM Static Random Access Memory

SRB Software Review Board

SRC Specimen Research Centrifuge

SREM Software Requirements Engineering Methodology

SRI Stanford Research Institute

SRM&QA Safety, Reliability, Maintainability, and Quality Assurance

SRMS Shuttle Remote Manipulator System

SRR System Requirements Review

SS Space Station -

SSA Structural Systems Analysis

SSA S-band Single Access

SSCB Space Station Control Board

SSCC Station Station Communication Center

SSCR Support Software Change Request
SSCS Space Station communication system

.....

SSCTS Space Station communications and tracking system

SSDMS Space Station data management system

SSDR Support Software Discrepancy Report

SSDS Space Station data system

SSE Software Support Environment

SSEF Software Support Environment Facility

SSIS Space Station Information System

SSME Space Shuttle Main Engine

SSO Source Selection Official

SSOCC Space Station Operations Control System

SSOCC Space Station Operations Control Center

SSOL Space Station Operation Language

SSON Spacelab Software Operational Notes

SSOS Space Station Operating System

SSP Space Station Program

SSPE Space Station Program Element

SSPO Space Station Program Office

SSSC Space Station Standard Computer

SSST Space Station System Trainer

STAR Self Test and Recovery (repair)

STARS Software Technology for Adaptable and Reliable Software

STDN Standard Number

STI Standard Technical Institute

STO Solar Terrestrial Observatory

STS Space Transportation System

SUSS Shuttle Upper Stage Systems

SYSREM System Requirements Engineering Methodology

Si Silicon

SubACS Submarine Advanced Combat System

TAI International Atomic Time

TBD To Be Determined

TBU Telemetry Buffer Unit

TC Telecommand

TCP Transmissions Control Protocols

TCS Thermal Control System

TDASS Tracking and Data Acquisition Satellite System

TDM Technology Development Mission
TDMA Time-Division Multiple Access

TDRS Tracking and Data Relay Satellite

TDRSS Tracking and Data Relay Satellite System

TFEL Thin Film Electrolumenescent

THURIS The Human Role in Space (study)

TI Texas Instruments

TM Technical Manual

TM Thematic Mapper

TMDE Test, Measurement, and Diagnostic Equipment

TMIS Technical and Management Information System

TMP Triple Multi-Processor

TMR Triple Modular Redundancy

TMS Thermal Management System

TPWG Test Planning Working Group

TR Technical Requirement

TRAC Texas Reconfigurable Array Computer

TRIC Transition Radiation and Ionization Calorimeter

TSC Trade Study Control

TSIP Technical Study Implementation Plan

TSP Twisted Shielded Pair

TSS Tethered Satellite System

TT&C Telemetry, Tracking, and Communications

TTC Telemetry Traffic Control

TTR Timed Token Ring

TWT Traveling-Wave Tube

U Non-restrictive

UCC Uniform Commercial Code

UDRE User Design Review and Exercise

UIL User Interface Language

UON Unique Object Names

UPS Uninterrupted Power Source

URN Unique Record Name

UTBUN Unique Telemetry Buffer Unit Name

UTC Universal Coordinated Time

V&V Validation and Verification

VAFB Vandenberg Air Force Base

VAX Virtual Address Exchange

VHSIC Very High-Speed Integrated Circuit

VLSI Very Large-Scale Integration

VLSIC Very Large-Scale Integrated Circuit
VV&T Validation, Verification and Testing

WAN Wide Area Network

WBS Work Breakdown Structure
WBSP Wideband Signal Processor

WDM Wavelength Division Multiplexing

WP Work Package

WRO Work Release Order

WS Workstation

WSGT White Sands Ground Terminal

WTR Western Test Range

XDFS XEROX Distributed File System
YAPS Yet Another Production System

ZOE Zone Of Exclusion
ZONC Zone Of Non-Contact

ZnS Zinc Sulfide

SSDS A/A STUDY PROGRAM PLAN REPORT

1.0 INTRODUCTION

The McDonnel Douglas Astronautics Company has been engaged in a Space Station Data System (SSDS) Analysis/Architecture A/A Study for the National Aeronautics and Space Administration, Goddard Space Flight Center. This Systems Engineering study which provides SSDS On-Board and Ground segment "strawman" designs within an end-to-end definition is divided into the following six tasks:

Task 1 - Requirements Definition

Task 2 - Options Development

Task 3 - Trade Studies

Task 4 - System Definition

Task 5 - Program Plan

Task 6 - Study Maintenance

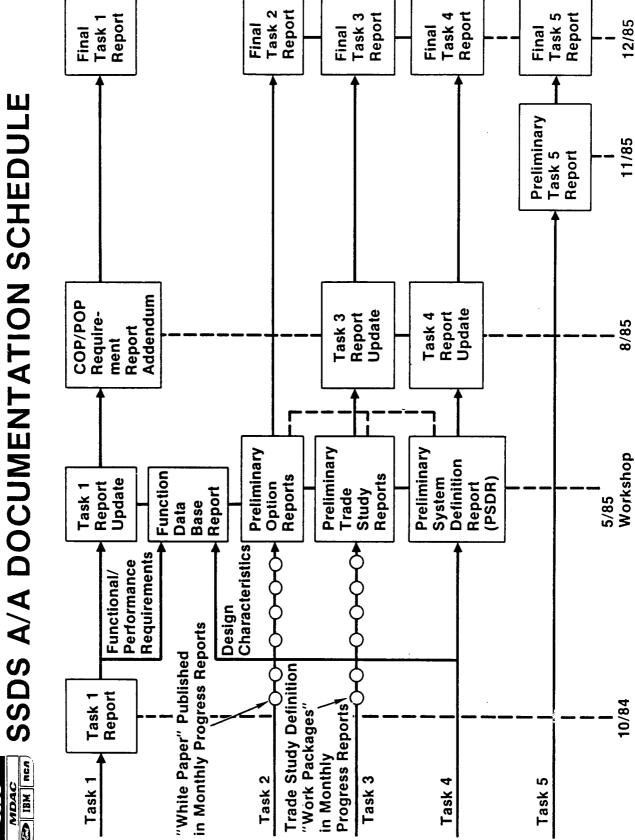
The preliminary Task 1 Report was published for NASA review in October, 1984. With the concurrency of the SSDS A/A Study and the Space Station Level B activities in the second quarter of 1985, NASA elected, by Contract Modification, to accelerate the study schedule to deliver preliminary Task 2, 3 and 4 Reports (plus an update to the Task 1 Report) in May, 1985, to support the JSC SSIS Workshop.

The Task 1, 3 and 4 Reports were revised and re-issued in August, 1985 based on comments generated from NASA/Industry review of the preliminary reports and from Quarterly Review presentations. Task 5 preliminary report of October 1985 was reviewed by NASA/Industry, resulting updates and expansions of cost backup have been included in this submittal.

This report contains the final output of the Program Plan (Task 5) effort. The report provides NASA with SSDS Program Schedule, Program Costs, and Advanced Technology Development recommendations as detailed in Sections 2, 3, and 4 respectively.

A summary of the SSDS A/A Study Report publication schedule is provided in Figure 1.1.





2.0 Executive Summary

2.1 Introduction

The original scope of the Task 5 Program Plan defined in the contractual Statement Of Work included detailed cost estimates and schedules, along with comprehensive program management and procurement strategies for the implementation of the complete system. The Program Plan scope was reduced, however, as part of the over-all study adjustments to accelerate the Task 2, 3, and 4 Reports in support of the Johnson Space Center SSIS Workshop, May 1985. As a result, the Program Plan, was re-defined to consist of the following efforts:

- o SSDS cost estimates (both On-Board and Ground segments)
- o Summary program schedules
- o Advanced Technology Development Recommendations

The SSDS elements addressed within these efforts are the On-Board and Ground system definitions provided within Section 6.0 and 7.0 respectively of the SSDS A/A Study Task 4 Report.

The efforts itemized above are discussed in detail in Sections 3.0, 4.0 and 5.0 of this report; it is the intent of this section to summarize the discussions and conclusions of those sections.

2.2 SSDS Summary Costs

Figure 2.2-1 provides a graphic over-view of the Work Breakdown Structure (WBS) developed for this Program Plan. Costs have been generated to the lower levels of this WBS but a generally reported only at the higher levels. 'Below the line' cost categories, Management, Systems Engineering and Integration SE&I, etc. have been collected against both On-Board and Ground segments and are so reported within the cost summaries.

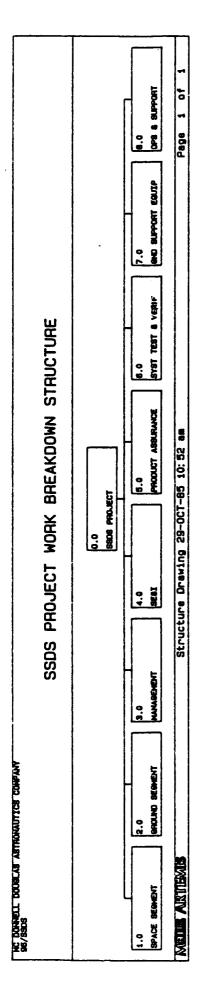


Figure 2.2-1

Detailed ground rules and assumptions for the SSDS cost estimates are provided in Section 3.0 and will not be duplicated here except to note that:

- o Implementation/operational costs for the Software Support Environment (SSE) and Development, Simulation, Integration and Training (DSIT) distributed capabilities are not included.
- o Only IOC costs are provided; growth is not addressed.
- o On-Board system implementation costs are provided through integration of SSDS elements into their associated launch packages, but launch and on-orbit 'build-up' support costs are excluded. No operational costs for the On-Orbit elements have been provided.
- o Operational costs for Ground facilities are limited to the staffing required to maintain the facilities; mission unique costs are not included.
- o The platform complement was limited to polar-orbiting platform, launch 1, POP1 deployment concurrent with Station IOC followed by POP2 then co-orbiting platform COP on six month centers.
- o Total hardware commonality was assumed between Station and platforms plus a high degree of software commonality.
- Cost methodology and cost data provided in this report represent engineering estimates and should not be construed as pricing data or used to develop pricing data.

2.2.1 Space Segment Cost Summaries

Costs for the Space Segment, WBS item 1.0, including H/W and S/W development, integration tests, and the recurring H/W production runs for the Space Station, POP1, POP2, and COP totalled \$445.4M; a breakdown of this cost into its component WBS elements is provided in Table 2.2.1-1.

Table 2.2.1-1

Space Segment Cost Summary

Integration Total (\$M)	21.5 311.9	.2 48.9	.2 42.3	1.2 42.3	.1 445.4
		-	-		25.1
Software <u>Development</u> (\$M)	145.1	15.4	8.8	8.8	178.1
Hardware Recurring (\$M)	82.1	32.3	32.3	32.3	179.0
Hardware <u>Development</u> (\$M)	63.2	0	0	0	63.2
Space Element	Space Station	POPI	P0P2	СОР	T0TALS (\$M)

2.2.2 Ground Segment Cost Summaries

WBS item 2.0 costs for the implementation of the nine ground facilities totalled \$402.36M; a breakdown of this cost is provided in Table 2.2.2-1.

2.2.3 Total SSDS Costs

The total SSDS costs, consisting of the all WBS element costs is \$1143.58M, representing \$339.02M recurring and \$804.56M non-recurring effort. A breakdown of these costs into the top level WBS elements is provided in Table 2.2.3-1.

2.2.4 Code S and Code T Funding

As part of the costing effort, NASA requested an estimate of funding responsibilities and associated SSDS costs for the Code S and Code T entities. The funding allocation, derived from NASA inputs and guidelines is summarized in Appendix A. The costs resulting from the allocation is:

Code S: \$814.9M Code T: \$328.7M

2.3 Program Schedule

The SSDS program scheduling effort has adopted milestones Contract Start Date (CSD), Preliminary Design Review (PDR), Critical Design Review (CDR), 1st Launch, and Initial Operating Capability (IOC), provided by NASA to the Level C/D activities. The Platform complement and deployment have not been as clearly defined therefore platform schedule assumptions have been made based on the Space Station Level C/D schedule and the Woods Hole update to the Langley Data Base. These assumptions are:

- o POP1 will be operationally deployed concurrent with Space Station IOC.
- o POP2 will be deployed six months after POP1.
- o The COP will be deployed six months after POP2.

Note that POP3, identified in the Langley Data Base, has been associated with the growth period and has not been addressed.

Table 2.2.2-1

Ground System Cost Summary

Facility	Hardware <u>Development</u> (\$M)	Hardware Recurring (\$M)	Software Development (\$M)	Integration (\$M)	Total (\$M)
Data Handling Ctr.	2	29.4	56	2.2	62.6
Level Zero Proc. Ctr 1	2.5	11.4	6	0.82	23.72
Level Zero Proc. Ctr 2	2	23.4	6	1.5	38.9
Level Zero Proc. Ctr 3	0	8.4	6	0.69	18.09
Ground Service Ctr	ιο.	2.88	5.59	0.34	8.82
SS Op Cont Ctr	2.5	27.7	49.3	3.08	82.58
SS Engr Data Ctr	1.2	16.9	30.5	1.9	50.57
P/F Cont Ctr	2.8	22.9	37.5	3.32	66.52
PF Engr Data Ctr	1.2	16.9	30.5	1.9	50.57
TOTALS (\$M)	20.2	160.02	206.39	15.75	402.36

Table 2.2.3-1

Total SSDS Cost Summary

Total (\$M)	20.6 634.6	508.98	20.6 1,143.58
GSE (\$M)	20.6	0	20.6
Syst Test & Verif. (\$M)	12.5	54.77	67.27
Product Assoc. (\$M)	7.2	1.87	6.07
Syst Engr & Integ. (\$M)	66.2	29.65	95.85
Mgmt (\$M)	82.7	20.33	40.85 103.03
Integ. (\$M)	25.1	60.02 206.39 15.75 20.33	40.85
S/W Devel (\$M)	1.871 0.67	206.39	339.02 384.49
H/W Recur (\$M)	179.0	160.02	339.02
H/W Devel (\$M)	63.2	20.2	83.4
Segment	Space	Ground	TOTALS(\$M) 83.4

Schedules for the Space and Ground segments are provided in Section 4.0. As noted in that section, the schedules are presented in simple bar chart form only; identification of key inter-relationships and critical paths were not addressed.

2.4 Advanced Development Recommendations

As part of the Program Plan, NASA requested advanced technology development recommendations addressing technologies utilized within the study strawman definitions that require advanced development but are not covered adequately within existing NASA sponsored activities. Section 5.4 lists technologies utilized directly within the strawman designs that will require demonstration prior to preliminary design decisions plus those technologies with high potential benefit to the SSDS development but which were discarded because of high development risk.

A compilation of the NASA advanced development plans was obtained from two sources: first and foremost was the Data Management System Advanced Development Project Plan; the second source was the Commerce Business Daily which was surveyed for supplemental NASA sponsored development.

The list of Study technologies was mapped onto the NASA advanced development plans and evaluated. The evaluation led to the recommendations that the following technologies by sponsored in further development:

- o Distributed Data Base Management
- o End-to-End Protocols/Formats
- o Command/Resource Management
- o Flight Qualified AI Machines

Additional details of the above subjects are provided in the appropriate section of this report.

3.0 Program Cost Estimates

3.1 Introduction

The SSDS is the combination of hardware and software that provides command and data management services to Space Station/Platform sub-systems and customers, both on orbit and on ground. Both basic segments, On-board and Ground have been addressed within the SSDS A/A Study and both have been cost . estimated as discussed in the following sections.

The Work Breakdown Structure defined for the Program Plan is presented and discussed in Section 3.3; estimated costs are consistent with that structure, however cost reporting is generally to the higher levels.

The costing effort Scope, Groundrules and Assumptions, presented in Sections 3.2 and 3.4 respectively, significantly bound and tailor the effort and should be carefully reviewed to fully understand the subsequent cost summaries. Methodologies utilized in Space and Ground segment estimates are provided in Section 3.5.

Finally, the cost summaries are provided in Section 3.6. A sub-section is provided for each segment followed by summaries of the total system. The latter summaries provide traditional 'non-recurring' vs 'recurring' break-out and estimates of NASA Code S and Code T funding responsibilities.

The cost methodology and cost data provided in this section represent 'engineering' estimates and should not be construed as pricing data or be used to develop pricing data.

3.2 Scope

Cost estimates are provided for the implementation (hardware and software) and operation of the SSDS Space and Ground System Definitions as provided in the SSDS A/A Task 4 Report. Elements defined to be outside the contractual scope of the Study, i.e. unique subsystem/user functions, Communications & Tracking, Audio/Video, Ground Distribution Networks, etc. are excluded from this effort.

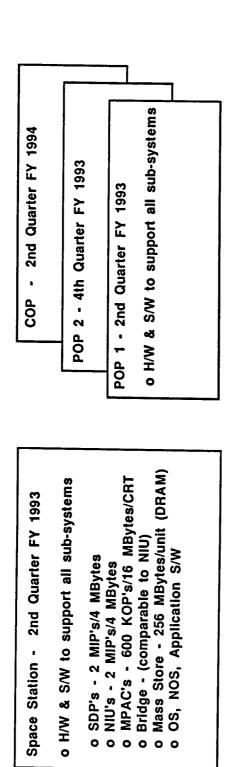
Growth phase design/implementation was not defined within the Study and will not therefore be estimated. Ground system operation costs and On-Board system spares have been provided for the ten (10) years following IOC and will be based on the IOC configuration.

The platform complement defined within the Woods Hole update to the Langley Mission Data Base provides two (2) polar orbiting platforms (POP's) and one (1) co-orbiting platform (COP) early in the initial operational period of the Space Station Program. A third POP is provided in the 1996 time frame. This last platform is considered to be part of the growth phase and has not been addressed.

The On-Board Systems definition for the cost effort is provided in Figure 3.2-1 and Table 3.2-1; the Ground System definition is provided in Figure 3.2-2 and Table 3.2-2. The facilities included in the pricing and a summary of the hardware items are shown.

The Software Support Environment tools, hardware and operation and the Development, Simulation, Integration & Training capabilities were not included in the cost estimates. These distributed capabilities were not sufficiently defined to be included in the Task 4 Report "strawman" designs and have been left to the Level B Contractor studies to design and cost. Simulation/Training hardware and software elements are also excluded along with ground versions of flight hardware.

The Onboard SSDS hardware and software components represent only a portion of the overall Space Station/Platform entities. It is therefore difficult, if not less than meaningful, to identify unique SSDS costs within the Launch and On-Orbit Assembly/Activation operations. Implementation costs of the On-Board system have been truncated at the completion of the integration of the SSDS elements into their appropriate Launch Packages. Launch and Build-up support on-board operational support and subsystem integration support have not been estimated; the recurring costs are limited to the hardware production run costs.



o 100 Mbps fiber optic global network

o Core and Payload sub-networks, bridge inter-connects

o ISO/OSI reference model and CCSDS protocols/formats incorporated

Total commonality between Station and Platform H/W

0

o High percentage commonality between Station and Platform S/W

On-Board SSDS System Definition Figure 3.2 - 1

<u>Table 3.2 - 1</u>

On-Board SSDS System Definition (Cost Model)

Space Station

o <u>Hardware Elements</u> :	SDP's	NIU's	MPAC's	BRIDGE's	MASS STOR'S
(Flt Units/Spares)	16/12	38/118	6/5	24/10	5/4

o Software Elements:

o Operating Systems:	Processor/MPAC OS	N/W OS
(KBytes/SLOC's)	160/40	410/102.5
o Applications:	Application Pqms	<u>Size</u> (KBytes/KSLOC's)
	Power	443/111
	GN&C	878/220
	Thermal	76/19
	ECLS	117/29
	Struct/Mech	108/27
	Crew	210/53
	Comm & Track	451/113
	Info & Data Mgmt	1628/407
	P/L Accom	466/117

<u>Table 3.2 - 1 - Cont'd</u>

On-Board SSDS System Definition (Cost Model)

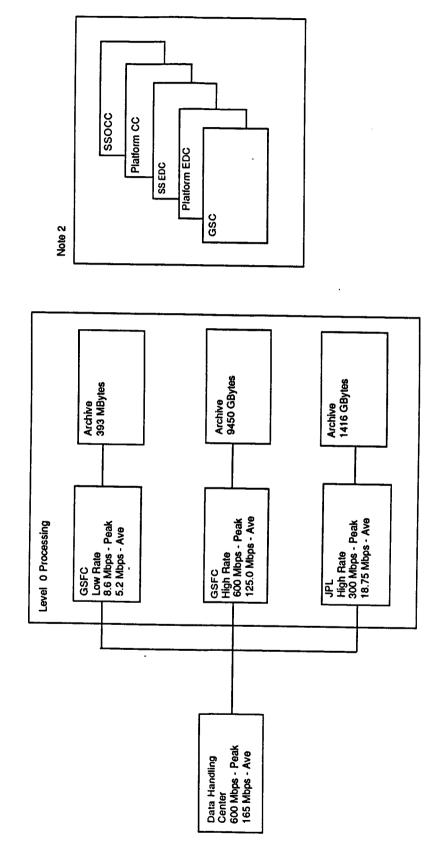
<u>Platforms</u>

o <u>Hardware Elements</u>: <u>SDP's NIU's BRIDGE's MASS STOR's</u> (Flt Units/Spares) 18/6 18/6 2/1 3/1

o <u>Software Elements</u>: Note - Figures represent modification effort of Space Station baseline software

o <u>Operating Systems</u>: <u>SDP OS</u> <u>N/W OS</u> (KBytes/KSLOC's) 20/5 50/12.5

o Applications: 490 KBytes/126 KSLOC's



Note 1 - Ground communications definition and cost not included in SSDS A/A Study Note 2 - For full Ground SSDS System connectivity, refer to Task 4, Section 7.0

Figure 3.2 - 2 Ground SSDS System Definition (Cost Model)

GROUND SSDS SYSTEM COMPATIBILITY/COST EQUIVALENTS

ITEM	MASS	OPTICAL	MAG.	LAN5	LANG	I AN	AN		MORK
ELEMENT	STORE	DISK	DISK ³			(MS)	(MF)	CONSOL	STATION4
ОНС	4								
LZPF (GSFC LOW)		-			_				
LZPF (GSFC HI)		-			_				
LZPF (JPL HI)		-			_				
6SC ²			20						
SSOCC			က			57	9		55
SS EDC		_	_				œ		
POP/COP CC			က			22	9		21
POP/COP EDC		-	-				80		
TOTAL	4	5	28	l	8	19	28	_	76

NOTES:

Line outage magnetic tape medium
For GSC similar to DEC RA81
Jisk storage similar to IBM 3880/3380
Workstation similar to Masscomp MC550
Similar to ETHERNET

Table 3.2-2 (cont'd) GROUND SSDS SYSTEM CAPABILITY/COST EQUIVALENTS

ITEM ELEMENT	PROCESSOR (4)	1 1	PROCESSOR PROCESSOR (1) (2)	PROCESSOR (3)	BRIDGE (MCC)	GATEWAY (FRONT END)	GATEWAY (LOCAL NET)	GATEWAY (DDN)	GATEWAY (TO EACH) ELEMENT)
рнс		3				-			
LZPF (GSFC LOW)		4				-			
LZPF (GSFC HI)		4		-				-	
LZPF (JPL HI)		⋖						-	
esc		ဗ	_						
SSOCC	ъ		7		^	_	c		9
SS EDC	4				ı	-	v (
P0P/C0P CC	က		2			-	v ~		
POP/COP EDC	4						8		
TOTAL	14	18	13	-	2	3	8	4	9

NOTES: 1 Similar to VAX 8600 cluster
2 LISP processor similar to Symbolics 3670
3 General processor similar to CYBER or CRAY
4 Similar to IBM 3083

Ground operation costs are limited to the staffing/operations required to maintain each facility; mission unique requirements are not estimated.

Foreign involvment has not been considered in SSDS acquisition and operations costing

Customer generated income from SSDS Services will not be estimated

Periods of performance for the cost estimation have been extracted from the schedules, also based on the WBS, provided in Section 4.0.

Total costs and time-phased distributions (by government fiscal year) are summarized in Section 3.6 in the following formats:

- o On-Board and Ground System costs
- o Full system costs
- o Non-Recurring vs Recurring Costs
- o Code T and Code S cost responsibilities

In all the above costs are provided in 1987 dollars, (escalated from 1985 dollars by a factor of 1.124).

3.3 Work Breakdown Structure (WBS)

The following Work Breakdown Structure has been generated for the SSDS A/A Study Program Plan Cost and Schedule Effort. Table 3.3-1 provides the WBS in tabular form; Figure 3.3-1 provides the WBS in graphic form. Table 3.3-2 provides the WBS Dictionary.

Table 3.3-1

SSDS A/A STUDY PROGRAN PLAN WORK BREAKDOWN STRUCTURE

0.0 SSDS

- 1.0 SPACE SEGMENT
- 1.1 SPACE STATION
- 1.1.1 HARDWARE
- 1.1.1.1 PROCESSORS
- 1.1.1.2 NETWORK INTERFACE UNITS (NIU'S)
- 1.1.1.3 BRIDGES
- 1.1.1.4 GATEWAYS
- 1.1.1.5 LOCAL AREA NETWORK
- 1.1.1.6 MASS STORAGE DEVICES
- 1.1.1.7 WORKSTATIONS
- 1.1.2 SOFTWARE
- 1.1.2.1 OPERATING SYSTEM
- 1.1.2.2 APPLICATION
- 1.1.2.3 MAINTENANCE
- 1.1.3 INTEGRATION
- 1.2 CO-ORBITING PLATFORM
- 1.2.1 HARDWARE
- 1.2.1.1 PROCESSORS
- 1.2.1.2 NIU'S
- 1.2.1.3 BRIDGES
- 1.2.1.4 GATEWAYS
- 1.2.1.5 LOCAL AREA NETWORK
- 1.2.1.6 MASS STORAGE DEVICES
- 1.2.2 SOFTWARE
- 1.2.2.1 OPERATING SYSTEM
- 1.2.2.2 APPLICATION
- 1.2.3 INTEGRATION
- 1.3 POLAR ORBITING PLATFORM
- 1.3.1 HARDWARE
- 1.3.1.1 PROCESSORS
- 1.3.1.2 NIU'S
- 1.3.1.3 BRIDGES
- 1.3.1.4 GATEWAYS
- 1.3.1.5 LOCAL AREA NETWORK
- 1.3.1.6 MASS STORAGE DEVICES
- 1.3.2 SOFTWARE
- 1.3.2.1 OPERATING SYSTEM
- 1.3.2.2 APPLICATION
- 1.3.3 INTEGRATION

2.0 GROUND SEGMENT

- 2.1 DATA HANDLING CENTER
- 2.1.1 HARDWARE
- 2.1.1.1 PROCESSORS
- 2.1.1.2 NIU'S
- 2.1.1.3 BRIDGES
- 2.1.1.4 GATEWAYS
- 2.1.1.5 LOCAL AREA NETWORKS
- 2.1.1.6 LOCAL BUSSES
- 2.1.1.7 MASS STORAGE DEVICES
- 2.1.1.8 WORKSTATIONS
- 2.1.2 SOFTWARE
- 2.1.2.1 OPERATING SYSTEMS
- 2.1.2.2 APPLICATION
- 2.1.3 FACILITY INTEGRATION & ACTIVATION
- 2.2 LEVEL ZERO PROCESSING FACILITY (LZPF) GODDARD SPACE FLIGHT CENTER (GSFC) LOW-RATE
- 2.2.1 HARDWARE
- 2.2.1.1 PROCESSORS
- 2.2.1.2 NIU'S
- 2.2.1.3 BRIDGES
- 2.2.1.4 GATEWAYS
- 2.2.1.5 LOCAL AREA NETWORKS
- 2.2.1.6 LOCAL BUSSES
- 2.2.1.7 MASS STORAGE DEVICES
- 2.2.1.8 WORKSTATIONS
- 2.2.2 SOFTWARE
- 2.2.2.1 OPERATING SYSTEMS
- 2.2.2.2 APPLICATION
- 2.2.3 FACILITY INTEGRATION
- 2.3 LEVEL ZERO PROCESSING FACILITY GSFC HIGH-RATE
- 2.3.1 HARDWARE
- 2.3.1.1 PROCESSORS
- 2.3.1.2 NIU'S
- 2.3.1.3 BRIDGES
- 2.3.1.4 GATEWAYS
- 2.3.1.5 LOCAL AREA NETWORKS
- 2.3.1.6 LOCAL BUSSES
- 2.3.1.7 MASS STORAGE DEVICES
- 2.3.1.8 WORKSTATIONS
- 2.3.2 SOFTWARE
- 2.3.2.1 OPERATING SYSTEMS
- 2.3.2.2 APPLICATION
- 2.3.3 FACILITY INTEGRATION

- 2.4 LEVEL ZERO PROCESSING FACILITY JET PROPULSION LABORATORY (JPL) HIGH-RATE
- 2.4.1 HARDWARE
- 2.4.1.1 PROCESSORS
- 2.4.1.2 NIU'S
- 2.4.1.3 BRIDGES
- 2.4.1.4 GATEWAYS
- 2.4.1.5 LOCAL AREA NETWORKS
- 2.4.1.6 LOCAL BUSSES
- 2.4.1.7 MASS STORAGE DEVICES
- 2.4.1.8 WORKSTATIONS
- 2.4.2 SOFTWARE
- 2.4.2.1 OPERATING SYSTEMS
- 2.4.2.2 APPLICATION
- 2.4.3 FACILITY INTEGRATION
- 2.5 GROUND SERVICES CENTER
- 2.5.1 HARDWARE
- 2.5.1.1 PROCESSORS
- 2.5.1.2 NIU'S
- 2.5.1.3 BRIDGES
- 2.5.1.4 GATEWAYS
- 2.5.1.5 LOCAL AREA NETWORKS
- 2.5.1.6 LOCAL BUSSES
- 2.4.1.7 MASS STORAGE DEVICES
- 2.5.1.8 WORKSTATIONS
- 2.5.2 SOFTWARE
- 2.5.2.1 OPERATING SYSTEMS
- 2.5.2.2 APPLICATION
- 2.5.3 FACILITY INTEGRATION
- 2.6 SPACE STATION OPERATIONAL CONTROL CENTER
- 2.6.1 HARDWARE
- 2.6.1.1 PROCESSORS
- 2.6.1.2 NIU'S
- 2.6.1.3 BRIDGES
- 2.6.1.4 GATEWAYS
- 2.6.1.5 LOCAL AREA NETWORKS
- 2.6.1.6 LOCAL BUSSES
- 2.6.1.7 MASS STORAGE DEVICES
- 2.6.1.8 WORKSTATIONS
- 2.6.2 SOFTWARE
- 2.6.2.1 OPERATING SYSTEMS
- 2.6.2.2 APPLICATION
- 2.6.3 FACILITY INTEGRATION

- 2.7 SPACE STATION ENGINEERING DATA CENTER
- 2.7.1 HARDWARE
- 2.7.1.1 PROCESSORS
- 2.7.1.2 NIU'S
- 2.7.1.3 BRIDGES
- 2.7.1.4 GATEWAYS
- 2.7.1.5 LOCAL AREA NETWORKS
- 2.7.1.6 LOCAL BUSSES
- 2.7.1.7 MASS STORAGE DEVICES
- 2.7.1.8 WORKSTATIONS
- 2.7.2 SOFTWARE
- 2.7.2.1 OPERATING SYSTEMS
- 2.7.2.2 APPLICATION
- 2.7.3 FACILITY INTEGRATION
- 2.8 POP/COP CONTROL CENTER
- 2.8.1 HARDWARE
- 2.8.1.1 PROCESSORS
- 2.8.1.2 NIU'S
- 2.8.1.3 BRIDGES
- 2.8.1.4 GATEWAYS
- 2.8.1.5 LOCAL AREA NETWORKS
- 2.8.1.6 LOCAL BUSSES
- 2.8.1.7 MASS STORAGE DEVICES
- 2.8.1.8 WORKSTATIONS
- 2.8.2 SOFTWARE
- 2.8.2.1 OPERATING SYSTEMS
- 2.8.2.2 APPLICATION
- 2.8.3 FACILITY INTEGRATION
- 2.9 POP/COP ENGINEERING DATA CENTER
- 2.9.1 HARDWARE
- 2.9.1.1 PROCESSORS
- 2.9.1.2 NIU'S
- 2.9.1.3 BRIDGES
- 2.9.1.4 GATEWAYS
- 2.9.1.5 LOCAL AREA NETWORKS
- 2.9.1.6 LOCAL BUSSES
- 2.9.1.7 MASS STORAGE DEVICES
- 2.9.1.8 WORKSTATIONS
- 2.9.2 SOFTWARE
- 2.9.2.1 OPERATING SYSTEMS
- 2.9.2.2 APPLICATION
- 2.9.3 FACILITY INTEGRATION
- 3.0 MANAGEMENT
- 3.1 SPACE SEGMENT
- 3.2 GROUND SEGMENT

- 4.0 SYSTEM ENGINEERING & INTEGRATION (SE&I) 4.1 SPACE SEGMENT
- 4.2 GROUND SEGMENT
- 5.0 PRODUCT ASSURANCE
- 5.1 SPACE SEGMENT
- 5.2 GROUND SEGMENT
- 6.0 SYSTEM TEST & VERIFICATION
- 6.1 SPACE SEGMENT
- 6.2 GROUND SEGMENT
- 7.0 GROUND SUPPORT EQUIPMENT (GSE)
- 8.0 OPERATIONS & SUPPORT

Figure 3.3-1

27

Table 3.3-2

WBS DICTIONARY

X.1 HARDWARE (General): Contract End Item (Black Box) costs

Includes:

- o DDT&E Costs
- o Production (& Spares) costs
- X.X.1.1 PROCESSORS: All SSDS processors used on-orbit or on-ground
- X.X.1.2 NIU's: All network interface units used on-orbit on on-ground
- X.X.1.3 BRIDGES: All bridges used to interconnect SSDS Local Area Networks (LAN's)
- X.X.1.4 GATEWAYS: All SSDS gateways
- $\rm X.X.1.5\ LOCAL\ AREA\ NETWORKS:\ All\ hardware\ associated\ with\ the\ on-board\ and\ ground\ LAN's$
- X.X.1.6 LOCAL BUSSES: All local (back end) and point-to-point bussing used in on-board and on-ground systems
- X.X.1.7 MASS STORAGE DEVICES: All mass storage devices and associated support hardware
- X.X.1.8 WORKSTATIONS: All workstations and associated hardware, on-board and on-ground
- X.2 SOFTWARE (General): Software costs will include:
 - o Design, Development, Coding, Verification & Validation
- X.X.2.1 OPERATING SYSTEMS; All Network Operating System (NOS) and Operating Systems (OS) software
- X.X.2.2 APPLICATION: All other non-NOS and OS software
- X.X.3 INTEGRATION: All efforts to integrate the SSDS within Space Station Program Element (SSPE)

3.0 MANAGEMENT

The management task consists of Business/Technical Mgmt, Configuration Mgmt, and Data/Information Mgmt. The Business/Technical Mgmt effort is required to provide services of Project Manager's staff, configuration management prepares and maintains the Configuration Management Plan and implementing procedures. The Data/Information Management is that effort required to assure that all responsibilities for preparation of contracturally required deliverable documentation/data/information are defined, assigned, scheduled, and statused; data needs are met.

4.0 SYSTEM ENGINEERING AND INTEGRATION

This task consists of the efforts required to perform and direct the system engineering, analyses, and integration activities in support of the design and development activities.

5.0 PRODUCT ASSURANCE

This element contains the effort required for the planning, conduct and support of the quality assurance, reliability, and safety and maintainability programs for both On-Board and Ground System.

6.0 SYSTEM TEST & VERIFICATION

Includes test equipment and operations by which the prime contractor will integrate, test and verify On-Board and/or Ground segment system level hardware and software.

7.0 GROUND SUPPORT EQUIPMENT

Includes the labor and material required to design, develop, manufacture, procure, assemble, test, checkout and deliver all of the various GSE hardware required for On-Board system checkout and final assembly. The GSE requirements include the equipment needed for test and checkout, fault isolation, handling and transportation.

8.0 OPERATIONS AND SUPPORT

Includes all ground efforts to support core system and maintain facility operations. Includes all non-mission unique activities. Activities include operations planning, scheduling, core systems monitoring, engineering data processing and archiving, etc.

3.4 Cost Estimation Ground Rules and Assumptions

In developing detailed and summary costs, the following ground-rules and assumptions have been adopted.

- o Contractor fees and NASA "wrap" costs have not been estimated.
- o Acquisition, modification, and operational costs for Institutional Facilities outside the SSDS boundaries are excluded.
- o Operational costs for existing and/or currently planned NASA Test Beds, Simulation Systems and Facilities are excluded.
- o H/W and S/W products have been considered as new development for the On-Board segment.
 - o All costs are expressed in FY 1987 dollars.
 - o Facility brick and mortar costs are excluded.
- o Application software for both Space and Ground segments will be estimated.
- o Hardware has been assumed to be common between Space Station and platforms (both COP and POP), thus eliminating development costs for platform equipment. Radiation hardening costs for the polar platform were not considered.
 - o Operational (Ground segment) costs are provided in man-years/year.
- o Recurring hardware (flight units/spares production) will be produced at cost effective production rate.

3.5 Costing Methodologies

As noted in the introduction to Section 3.0, the cost estimations provided here-in have been somewhat coarse due to schedule and resource limitations. Costs have been generally based on empirical approximations, end-item comparisons, and engineering estimates. Identified methodologies are listed in the following two sections addressing Space segment and Ground segment estimates respectively.

3.5.1 Space Segment Methodologies

- o Spares Requirements: The required number of spares was computed by assuming:
 - 1) a re-supply interval of 3 months for the Space Station and a 6 month interval for the platforms.
 - 2) a unit MTBF of 10,000 hours when powered and 100,000 when not powered.
 - 3) redundant platform elements are powered off until needed.
 - 4) a 99% probability that the minimum required hardware complement is available.

o Software Sizing: Memory sizing was ignored when identified as buffer storage (C&T-400 KBytes, IDMS-950 KBytes, Cust. P/L-2MBytes). Bytes were converted to source lines of code (SLOC) by the factor 4 bytes/SLOC. Because of the enforcement of commonality, the size of the unique S/W for POP1 was assumed to be 10% of the Space Station S/W size and 5% was assumed for POP2 and COP.

o System Test and Verification: Costed as a 10 man level of effort (LOE) for each STS launch package; 3 concurrent events for the Space Station and 1 event for each platform. Application S/W support was assumed to be provided by the subsystem contractor.

- o Hardware Elements: The Subsystem data Processor (SDP), Network Interface Unit (NIU), and Bridge (B) elements were considered to be of equal complexity and cost; their cost was based on the Orbiter upgrade computer. The non-recurring cost for these components was based on the modification effort for a comparable unit being developed for another program. The Workstation (WS) and Mass Storage (MS) costs were based on Orbiter and other current program experience.
- o Ground Support Equipment: The GSE requirements were based on the need to support four facilities and production simultaneously. Unique equipment was assumed for each hardware element with a total of 8 each for the SDP, NIU and B, and 6 each for the MS and WS.
- o Software Development/Maintenance: OS and NOS software development was costed using a 65 SLOC/man-month productivity factor over a three year period. Follow-on maintenance until launch utilized a productivity of 6000 SLOC/man-month LOE. Development and maintenance factors utilized for application software were 110 SLOC/man-month and 15000 SLOC/man-month LOE respectively. These factors are based on STS program experience.
 - o Labor: Labor was estimated at \$105,000/yr-man, (1985).
- o Data Management System (DMS) Integration: DMS integration costs for the Space Station include a facilities cost of \$16M (1985) and a 20 man LOE; each platform cost consisted only of a 10 man LOE.
- o Product Assurance: Product Assurance was estimated as 4% of the recurring hardware cost. Software QA costs are included in the productivity factors.
- o System Engineering and Integration/Management: SE&I costs were estimated as 15% of the S/W, H/W and GSE costs; management (includes project and line management) costs were estimated as 15% of all other costs.

3.5.2 Ground Segment Methodologies

- o Design, development, testing and engineering (DDT&E): DDT&E for Commercial Off-The-Shelf (COTS) equipment are zero; DDT&E costs for special-build hardware were not shared between facilities for similar equipment.
- o Hardware Procurement: All hardware was considered to be purchased, not leased; no volume purchase discounts were considered.
 - o Application Software: Application software costs are 3 to 4 times the cost of the associated processor hardware.
 - o Software Commonality: The re-use of applications software between a Space Station and a Platform facility was not assumed.
 - o Application Software Maintenance: Annual applications software maintenance costs are 20% of the development costs and are not shown.
 - o Facility Integration: Facility Activation and System Integration were estimated as 15% of the total H/W and S/W costs for the facility.
 - o Facility Operational Staffing: It was assumed that key facility positions were continuously manned. It is recognized that if scheduling of mission activities was optimized with respect to the ground controller crew, the number of controllers could be reduced. This was not considered likely or desirable however, due to scheduling complications and the requirement for contingency support.

3.6 Cost Summaries

3.6.1 Space Segment

The total cost for implementation of the Space Segment, as bounded by the estimation scope, ground rules and assumptions is \$634.6M. In summary this figure represents the implementation cost to develop and support the SSDS Space Segment through integration into the appropriate launch packages; costs beyond that point, i.e. for support of launch, build-up support and subsequent operations are not included. It is noted, however, that some (pre-IOC) software maintenance for both OS/NOS and application programs is included in these estimates.

These costs are all non-recurring with the exception of hardware production run costs. Summation of these costs yields a distribution of \$160.1M Recurring, and \$474.5M Non-Recurring.

Cost estimates for the individual WBS elements are provided in Table 3.6.1-1.

Details of the hardware development/recurring costs and software sizing/development costs for both the Space Station and platform(s) are provided in Figures 3.6.1-1 through 3.6.1-6.

Time-phased distributions of these elements are provided in Figures 3.6.1-7 through 3.6.1-15.

Table 3.6.1-1
SPACE SEGMENT SUMMARY COSTS

WBS	DESCRIPTION	RECURRING (\$M)	NON-RECURRING (\$M)
1.1	Space Station	63.2	248.7
1.2	COP	32.3	10.6
1.3	POP 1	32.3	16.6
1.3	POP 2	32.3	10.0
3.1	Management	0	82.7
4.1	SE&I	0	66.2
5.1	Product Assurance	0	7.2
6.1	System Test & Verif	0	12.5
7.1	Ground Support Equipment	0	20.6
	TOTALS:	160.1	474.5

Figure 3.6.1-1

SPACE STATION HARDWARE COSTS

HARDWARE ELEMENTS	Development Cost (\$M)	Recurring Cost(\$M)/ Flight Unit	Total Flight Units	Fl.Units Recurring Cost(\$M)	Total Spare Units	Spares Recurring Cost(\$M)	Total Recurring Cost(\$M)	Total Cost(\$M)
SUBSYS.DATA PROC	22.5	0.562	16	9.0	12	6.7	15.7	38.2
NETWORK INT.UNIT	22.5	0.562	38	21.4	18	1.01	31.5	54.0
WORKSTATION	7.0	0.703	9	4.2	2	3.5	7.7	14.8
MASS STORAGE	5.6	0.899	2	4.2	4	3.6	8.1	13.7
BRIDGE	5.6	0.562	24	13.5	30	5.6	19.1	24.7
TOTALS	63.2		68	52.5	49	29.6	82.1	145.4

Figure 3.6.1-2
SPACE STATION SOFTWARE COSTS

FUNCTION	Memory Size Software Size (K Bytes) (K SLOC)	Software Size (K SLOC)	Development Cost (\$M)	Maintenance Cost (\$M)	Total (\$M)
OPERATING SYSTEM	160	40	6.1	2.4	8.4
NETWORK OPERATING SYSTEM	410	102.5	15.5	6.0	21.6
T0TALS	570	142.5	21.6	8.4	30.0

Figure 3.6.1-3

SPACE STATION - APPLICATION SOFTWARE COSTS

FUNCTION	Memory (K Bytes)	Software S121ng (K SLOC)	Development Cost (\$M)	Maintenance Cost (\$M)	Total (\$M)
A SMUC JI ALL BUMER	443	111	6 6	1.1	11.6
GNAC	878	220	19.6	3.5	23.1
THERMAL CONTROL	9/	19	1.1	0.3	2.0
	711	29	2.6	0.5	3.1
STRUCTURES/MECHANISMS	108	7.2	2.4	0.4	2.8
CREW SYSTEMS	210	53	4.7	0.8	5.5
COMM & TRACKING	451	113	10.1	1.8	11.9
INFO & DATA MANAGEMENT	1628	407	36.4	6.4	42.8
PL & SERVICING ACCOM.	466	711	10.4	1.8	12.2
TOTALS	4377	1094	97.8	17.2	115.1

Figure 3.6.1-4

SPACE PLATFORM HARDWARE COSTS

Total Recurring Cost(\$M) 13.5 13.5 3.6 7.7 32.3 Spares Recurring Cost(\$M) 9.0 8.2 0.9 Total Spare Units စ 14 Fl.Units Recurring Cost(\$M) 10.1 2.7 24.1 10.1 Total Flight Units 18 18 ~ 4 Recurring Cost(\$M)/ Flight Unit 0.562 0.562 0.899 0.562 SUBSYS.DATA PROC NETWORK INT.UNIT MASS STORAGE HARDWARE ELEMENTS BRIDGE TOTALS

Figure 3.6.1-5

SPACE PLATFORM SOFTWARE COSTS (FIRST PLATFORM)

FUNCTION	Memory Size (K Bytes)	Memory Size Software Size Development (K Bytes) (K SLOC) Cost (\$M)	Development Cost (\$M)	Maintenance Cost (\$M)	Total (\$M)
OPERATING SYSTEM	20	5	0.8	0.2	1.0
NETWORK OPERATING SYSTEM	90	12.5	1.9	9.0	2.5
		*			***************************************
TOTALS	70	17.5	2.7	0.8	3.5

Figure 3.6.1-6
SPACE PLATFORM - APPLICATION SOFTWARE COSTS (FIRST PLATFORM)

FUNCTION	Memory Size (K Bytes)	Software Size (K SLOC)	Development Cost (\$M)	Maintenance Cost (\$M)	Total (\$M)
ELECTRIC POWER	20	13	1.1	0.10	1.2
GN&C	06	23	2.0	0.18	2.5
THERMAL CONTROL	10	က	0.2	0.05	0.5
ECLSS	50	5	0.4	0.04	0.5
STRUCTURES/MECHANISMS	50	5	0.4	0.04	0.5
CREW SYSTEMS	30	œ	7.0	0.06	0.7
COMM & TRACKING	20	13	٦.٦	0.10	٦.2
INFO & DATA MANAGEMENT (ID)	DM) 170	43	3.8	0.33	4.1
PL & SERVICING ACCOM. (PL&	<u>.</u>	13		0.10	1.2
TOTALS	490	123	11.0	96.0	11.9

NOTE: This figure represents the effort to modify the Space Station baseline software for platform utilization.

Figure 3.6.1-7. Space Station Cost Summary

MBS MD.	NBS NG. DESCRIPTION FY 1987 FY 1989 FY 1990 FY 1991 FY 1992 FY 1994	FY 1987	,		FY 1988	_		_	FY 1989			1	FY 1990			٤	FY 1991			FY 1992	442			FY 1993	993			FY 1994	994	i	
	30 40 10 20 30 40	\$	38 48 18 29 38	Ga	10 20 30	<u>.</u>		_	10 20 30 40	<u>,</u>	 	~	19 29 30	10 20 38 40 10 20 30 40 10 20 30 40 10 20 30 40 10 20 30 40 10 20 30 40	2	8	10 20 30 40	9	2	10 20 30 40	s	2	=	10 20 30	8	20 30 40 10 20 30 40	=	æ	10 20 30 40	2	101AL (SR)
	MILESTONES	653 +																			•	+ 1st Launch	=		<u> </u>	+ 10C, POP1 + POP2	÷	2	900		
===	STATION HARDWARE -NON-RECURING -RECURING	7.9 7.9 7.9 7.9 7.9 7.9 7.9 2.76 2.76 2.76 2.76 2.76	7.9 7	 2	7. 9.7	7. 2.		7.9	76 2.	76 2.	76 2.	78 2.1	76 5.ž	7.9 7.9 2.76 2.76 2.76 2.76 2.76 2.76 5.22 5.22 5.22 5.22 5.22 5.22 5.22 5.2	2 5.Z	2.2	5.22	5.22	5.22	5.22	2.44	2.46	2.46	2.48							63.2 82.1
1.1.2	STATTON SOFTWARE	1.79 1.79 1.79 1.79 9.94 9.94	79 1.	79 1.	79 9.	£		.4	9.	4.	*	94 9.	3.8	9,94 9,94 9,94 9,94 9,94 9,94 8.85 8.85 8.85 8.85 2.85 2.85 2.85 2.85	5.00	9.	2.85	2.85	2.85	2.85	2.85	2.85	2.85	2.85							145.1
1.1.3	INTEGRATION						'n	.S.	.S.	.∺ %	3.56 3.58 3.56 3.58 3.58 3.58	3.5	28																		21.5
2.4101	TOTALS BY EY BUIGBTEB 0.40 12.4 12.4 12.4 20.4 20.4	9.49 12.4 12.4 20.4 20.4	2	27	8	97	7	77	1 5	**	41 6	7	7	24. 24. 14.2 14.2 14.2 14.0 14.0 14.0 14.0 14.0 14.0 14.0 14.0			9 0	70 6	6	6	2.5	- 5	2.5	15	•	•	•	•	-	-	24. 24. 14.2 14.2 14.2 14.0 14.0 14.0 14.0 14.0 18.07
		:	: :	:	: :	!		:	! :	: :	! !	! :	: :	: !	:	:	;	; •	•	<u>;</u>	;	;	;	; ;	,	•	•	•	•	•	:

Figure 3.6.1-8. POP1 Cost Summary

	WES RU. DESCRIPTION				FY 1988	-			FY 1989	_		_	FY 1990			_	FY 1991	_		_	FY 1992			4	FY 1993			Ę	FY 1994		
	39 48 19 20 30 40	97 98	-	30 40 10 20 30	2	g	9	=	10 20 39		9	9	19 29 39 40	æ	9	2	10 20 30	9	9	3	10 20 30 40	[]	-	7	10 20 30	19 29 39 49 19 29 39 49 19 29 39 49 19 29 39 49 19 29 39 49	=	22	10 20 30 40	2	
=	HILESTONES	053 +																			•	+ ist Launch	unch		•	+ 10C,P0P1 + P0P2	=	70P2	3 03 +	3	
1.3.1 PD	POP HARDWARE (RECURRING DNLY)	46 DIR.Y)					-	.11	7.	71 1.	71 1.	71 1.	71 2.	39 2.	39 2.	39 2.	39 2.	1.71 1.71 1.71 1.71 1.71 1.71 2.39 2.39 2.39 2.39 2.39 2.39 2.39 2.39	39 2.	39 2.	39 0.	9	9.0	9.	9						
1.3.2 PB	POP SOFTWARE						•	E	33 1.	ب ا	24 I.	Z4 1.	24 1.	24 1.	24 0.	99 0.	99	0.33 0.33 1.24 1.24 1.24 1.24 1.24 1.24 0.99 0.99 0.99 0.99 0.99 0.99 0.12 0.32 0.32 0.32 0.32	99 0.	.0	99 0.	72 0.	12 0.3	12 0.3	22						
1.3.3 IN	INTEGRATION												ď	29 0.	0.29 0.29 0.29 0.29	28	23														

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Figure 3.6.1-9. POP2 Cost Summary

HBS NO.	MBS ND. DESCRIPTION FY 1987 FY 1994 FY 1994 FY 1994	FY 1987		FY 1988	FY 1988			2	FY 1989			FY 1990	FY 1990			Œ	FY 1991			F	FY 1992			FY 1993	FY 1993			FY 1994	-	;	
	39 40 10 20 30 40 10 20 30 40 10 20 30 40 10 20 30 40 10 20 30 40 10 20 30 40 10 20 30 40	30	9	20 20	25	2	=	20	25	10 20 39 40	9	25	Ħ	2	=	82	23	2	9	8	Ħ	#	9	8	8	10 20 30 40 10 20 30 40 10 20 30 40 10 20 30 40 10 20 30 40	=	10 20 30 40	97		101AL (\$M)
	MILESTONES	053 +																			+	+ 1st Launch	ŧ		70.	+ 100, P0P1 + P0P2	+ P0P2		+ COP		
1.3.1	POP HARDWARE (RECURRING DMLY)	(ME ONLY)							1.71	1.7	1.71	1.7	1.7	1.71	2.39	2.39	2.39	2.39	2.39	2.39	2.39	2.39	0.68	99.0	1.71 1.71 1.71 1.71 1.71 1.71 2.39 2.39 2.39 2.39 2.39 2.39 2.39 2.39	9.6B				. •	32.3
1.3.2	POP SOFTWARE								9.16	0.18	0.7	6.			6.7	0.7	0.56	6.56	9.56	6.5	0.5	0.56	0.17	0.17	0.18 0.18 0.7 0.7 0.7 0.7 0.7 0.7 0.5 0.56 0.56 0.56 0.56 0.55 0.17 0.17 0.17 0.17	0.17					89
1.3.3	INTEGRATION														0.29	0.29	0.29 0.29 0.29 0.29	0.29													1.2
																															1
TOTALS	TOTALS BY FY QUARTER	•	0	-	-	-	-	_	1.69	1.89	2.41	2.4	2.41	2.41	3.38	3.38	3,24	3.24	2.45	2.45	2.95	2.45	6.83	0.85	6.65	0 0 0 1.89 1.89 2.41 2.41 2.41 2.41 3.41 3.38 3.34 3.24 3.24 2.95 2.95 2.95 0.85 0.85 0.85 0.85 0.85 0 0 0 0 42.3	•	•	•	-	42.3

Figure 3.6.1-10. COP Cost Summary

KBS ND.	WES NO. DESCRIPTION FY 1987 FY 1988 FY 1989 FY 1989	FY 1987	_	F	FY 1988			1	FY 1989			3	FY 1990			EV 1991	106			FV 1997	600			£v 1007	2		1	1001			
						1	ł			;	i				į				!		****			24.7	,	'	-	1774			
		9 8	30 40 10 20	32	25	\$	=	25	10 20 30 40	\$	2	30	2	2	2	70	23	\$	2	30	2	7	2	20	19 29 30 48 19 20 39 40 19 20 30 40 19 20 30 49	<u>-</u>	10 20		30	101AL	-
																														£	_
	MILESTONES	CS3 +																			+ 151	+ 1st Launch	_		+ 10C, P0P1 + P0P2	÷	P0P2	•	e - CO -		
1.2.1	COP HARDWARE (RECURRING ONLY)	46 ONLY)									1.71	1.71	1.71	1.71	1.71	1.3	2.39	2.39	2.39	2.39	2.39	2.39	2.39 2	39 0	1.71 1.71 1.71 1.71 1.71 1.71 2.39 2.39 2.39 2.39 2.39 2.39 2.39 2.59 0.68 0.68 0.68 0.68	.0 89	.0	9		32.3	
1.2.2	COP SOFTWARE										0.18	0.18	0.7	. 0.7	6.7	0.7	0.7	0.7	0.56	.58	9.56	0.56	.56	35.	0.18 0.18 0.7 0.7 0.7 0.7 0.7 0.7 0.5 0.56 0.56 0.56 0.56 0.56 0.56 0.17 0.17 0.17 0.17	17 0.	17 0.	-1			
1.2.3	INTEGRATION																0.29	0.29 0.29 0.29 0.29	0.29	6.3										1.2	
							;																								
TOTALS !	TOTALS BY FY QUARTER	•	0 0 0			•			•	•	1.89	1.89	2.41	2.41	2.41	2.4	3.38	3.38	3.24	1.24	2.95	.95 2	.95 2	. 25	.0	85 0.0	65 9.	2		0 0 0 0 0 1.89 1.89 2.41 2.41 2.41 2.41 3.38 3.38 3.24 3.24 2.95 2.95 2.95 2.95 0.85 0.85 0.85 0.85 0 0 42.3	

Figure 3.6.1-11. Management

ON SEP	#BS NG. DESCRIPTION FY 1987 FY 1989 FY 1990 FY 1991 FY 1992 FY 1993 FY 1994	FY 1987		FY 1988	98			FY 1989	8			FY 1990	2			FY 1991			F	FY 1992			FY 1993	1993			FY 1994	*		
	36 46 16 26 36 46	30 40 10 29 30 40	=	10 20 30	25		10 20 39 46	20 30	10 20 30 40		9	10 20 30	10 20 30 40		7	10 20 30	#	; =	22	10 20 30 40	\$	2	8	10 20 30	.	19 ZB 39 40 10 20 30 40 1016LS	10 20 30	a	=	OTALS
	MILESTONES	053 +				,													! !	7	+ 1st Launch	4		-	C, P0P1	+ ist Launch + 10C,POP! + PDP2 + COP	2	8		<u> </u>
3.1.1	SPACE STATION	1.91 17.8 2.84 2.84 4.06 4.06	2.84	2.84	4.06	4.06	4 :	4.6	2.9	2.9	2.9	2.9	4.6 4.6 2.9 2.9 2.9 2.9 2.7 2.7 2.7 2.7 1.36 1.36 1.36 1.35 0.81 0.81 0.81 0.81	1.7 2	.7 2	.7 1.3	e	1.3	1.3	9.0	0.81	0.81	0.81							58.4
3.1.2	POP 1						0.39	0.39	0.53	3.53	.53 6	.53	0.39 0.39 0.53 0.53 0.53 0.53 0.67 0.67 0.68 0.68 0.64 0.64 0.64 0.16 0.16 0.16 0.16	67 0.	99	9.0	4 0.6	4.0.4	9.0	0.16	0.16	0.16	0.16							.
3.1.3	POP 2							-	3.36	35.0	÷.	÷.	0.36 0.36 0.44 0.44 0.44 0.44 0.58 0.58 0.56 0.56 0.56 0.56 0.56 0.58 0.18 0.14 0.14 0.14	3	8	8.0	4.0.4	- 0.S	9.5	6.56	0.56	9.18	0.14	÷.						1.1
3.1.4	800									-	.36 6	.36 0	0.34 0.34 0.44 0.44 0.44 0.44 0.58 0.58 0.61 0.61 0.56 0.56 0.56 0.56 0.14 0.14 0.14 0.14	÷	÷	H 0.5	8 0.5	9.0	0.6	0.56	9.56	0.56	0.56	1.	6.14	0.14	9.14			1.1
TOTALS	TOTALS BY FY QUARTER 1.91 17.8 2.84 2.84 4.06 4.06	1.91 17.8 2.84 2.84 4.06 4.06	2.84	2.84	30.	i .	66.	8-	1.79	.79	23	23.	6 4.99 4.99 3.79 4.23 4.23 4.25 4.25 4.4 4.4 3.14 3.19 3.17 3.17 2.09 2.09 1.71 1.67 6.28 6.28 6.14 6.14 0 0 82.7	g	-	3.5	3.1	3.1	3.17	2.09	2.09	1.31	1.67	0.28	0.28	0.14	6.14	•	-	

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Figure 3.6.1-12. System Engineering and Integration

NGS MU. DESLKIFIJUM											•	141			Ξ	14.1992			FY 1993	2		Œ	FY 1994	
	30 40	36 46 19 26 39	2	10 20 30 40	Ŗ	2	9	10 20 39	9	2	10 20 30	9	#	: =	20	10 20 30		2	10 20 30	19 20 30 40 10 20 30 40 10 20 30 40 10 20 30 40		25	10 20 30 40 TOTALS	; 3
MILESTONES	#11ESTONES + CSD + 10C,POP1 + PUP2 + COP															=	+ ist Launch	=		1,201 +	+ 10C, P0P1 + P0P2	P0P2	9	2
4.1.1 SPACE STATION		2.91 2.91 2.91 2.91 2.91 2.91		2.91 2.91 2.91 2.91 2.91 2.91 2.91 2.91	2.91	2.91	2.91	2.91	2.91 2	.91 2.	91 2.	16												
1.1.2 POP 1			0.51	0.51 0.51 0.51 0.51 0.51 0.51 0.51 0.51	0.51	0.51	0.51	0.51).51 ¢	.51	51 0.	51 0.	1 0.5	0.5	0.51									
4.1.3 PDP 2					0.4	₽.	÷.	9.44		÷.	0.44 0.44 0.44 0.44 0.44 0.44 0.44 0.44	44 0.	4.0	÷.0	9.4	0.4								
4.1.4 COP							.4	9.4). +	÷.	0.44 0.44 0.44 0.44 0.44 0.44 0.44 0.44	2.0	4.0	4.0.4	9.	9.	÷.	 *						

Figure 3.6.1-13. Product Assurance

NBS ND.	NBS ND. DESCRIPTION	FY 1987		FY 1988	88			FY 1989	686			Ž	FY 1990			2	FY 1991			Ξ.	FY 1992			Ξ	FY 1989 FY 1990 FY 1991 FY 1992 FY 1994			7	FY 1994		
		39 40 19 20	=	10 20 30	23	2	=	2	10 20 39 40	10 20 39 40 10 20 39 40	=	10 20 30	25	3	2	28	10 20 30	.	=	30	10 20 30 40 10 20 30 40	\$	=	20	10 20 30 40	.	. =	22	10 20 30 40	9	TOTALS
	H1LESTONES + CSD + 1st Launch + 19	050 +																			-	+ 1st Launch	5		1		•	240	C, POP1 + POP2 + COP	۰	Ê
5.1.1	SPACE STATION	0.13 0.13 0.13 0.13 0.13 0.13 0.13 0.13	0.13	0.13	0,13	0.13	0.13	0.13	0.13	0.13	0.13	6.13	0.13	0.13	0.13	0.13	0.13	0.13	0.13	0.13	0.13	0.13	0.13	0.13							3.3
5.1.2	POP 1						0.03	0.07	0.02	0.07	0.01	0.07	0.07	0.03	0.0	0.0	0.0	0.0	0.07	0.0	0.07	0.07	0.07	0.01 0.07 0.07 0.07 0.07 0.07 0.07 0.07							
5.1.3	P0P 2									0.07	0.0	0.07	0.07	0.0	0.0	0.07	0.0	0.07	0.07	0.07	0.0	0.0	0.0	0.01	0.01 6.07 6.07 6.07 6.07 6.07 6.07 6.07 6.07	0.0					1.3
5.1.4	&										0.0	0.0	0.01	0.0	0.01	0.03	0.07	0.01	0.0	0.07	0.07	0.07	6.07	0.07	0.01 0.01 0.01 0.01 0.01 0.01 0.01 0.01	0.07	0.0	0.07			1.3
TOTAL S.	TO SEA TO A SEA THE A SEA SEA SEA SEA SEA SEA SEA SEA SEA		1			-	5	5	3																1 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0						

Figure 3.6.1-14. System Test and Verification

NBS NO.	MBS NO. DESCRIPTION	FY 1987		FY 1988	FY 1988	:		FY 1989	686			FY 1990	9.		_	FY 1991	_		Œ	FY 1992			σ	FY 1993			Œ	FY 1994		
		39 49 18 29 39	=	25	8	2	2	10 20 30 40	æ	#	=	10 20 30	33	2	2	10 20 30	2	, gg	10 20 30	e e	¥ -	: 3	72	10 20 30	\$	9	10 20 30	g	#	10 20 30 40 10 20 30 40 10 20 30 40 10 20 30 40 10 20 30 40 10 20 30 40 107ALS
	H1LESTONES + CSD + 10C, PDP1 + PDP2 + CDP	0 50 +							i i i i i											*	+ 1st Launch	unch.		+	+ 10C, POP1 + POP2	=	20F2	a .	a	£
6.1.1	SPACE STATION											-	989.0	.88	85	8	88	.0	0.88 0.88 0.98 0.89 0.88 0.88 0.88	22										7.1
6.1.2	- 904													•	.29	.29	83	29 0.	0.29 0.29 0.29 0.29 0.29 0.29	œ.										1.9
6.1.3	POP 2																•	29 0.	0.29 0.29 0.29 0.29 0.29 0.29	6.	19 0.2	9 0.2	<u>.</u> ق							1.8
6.1.4	400																	•	0.29 0.29 0.29 0.29 0.29 0.29	 8:	19 0.2	9 0.2	9 0.2	9 -						æ:
TOTALS	TOTALS BY FY QUARTER 0 0 0 0 0 0	0 0 0 0	•	۰	•	•	•	0	•	•	•	. 0	.68	88	.1 1.	17 1.	1 1.	9	7.1 27	5 0.5	6 0.5	2.0	9 0.2					9	9	0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0

Figure 3.6.1-15. Ground Support Equipment

39 40 10 20 30 40 10 20 30 40 10 20 30 40 10 20 30 40 10 20 30 40 10 20 30 40 10 20 30 40 10 20 30 40 107ALS ### CAST	10 20 30 40 10 2				************
MILESTONES + CSD 3.42 3.42 3.42 3.42 3.42		A1 A4 AC A	20 30 40 10 2	. 30 40 10	30 40
SPACE STATION 3.42 3.42 3.42 3.42			+ 1st Launch	+ IDC,POP1 + POP2 + COP	6 53 +
7.1.2 POP 1					
2.1.3 P0P 2					
7.1.4 COP					

3.6.2 Ground Segment Costs

3.6.2.1 Implementation Costs

The implementation cost for the Ground Segment, again as bounded by the estimation scope, ground rules and assumptions, is \$509.01M. This figure represents the full development and integration costs for ground system readiness to support IOC. The estimates utilized a large percentage of Commercial-Off-The-Shelf (COTS) hardware which is categorized as "recurrent" costs; the total implementation cost breaks down to \$160.05M Recurring and \$348.96M Non-Recurring.

Costs for the individual WBS elements are provided in Table 3.6.2-1. Time-phased distributions are provided in Figure 3.6.2-1 through 3.6.2-22.

Table 3.6.2-1

Ground Segment Summary Costs

WBS	DESCRIPTION	RECURRING (\$M)	NON-RECURRING (\$M)
2.1	DATA HANDLING CENTER	29.4	33.2
2.2	LZPF GSFC LO-RATE	11.4	12.3
2.3	LZPF GSFC HI-RATE	23.4	15.5
2.4	LZPF JPL HI-RATE	8.4	9.7
2.5	GROUND SERVICES CENTER	2.88	5.94
2.6	SPACE STATION OPERATION CONTROL CENTER	27.73	54.88
2.7	SPACE STATION ENGINEERING DATA CENTER	16.97	33.60
2.8	PLATFORM CONTROL CENTER	22.9	43.62
2.9	POP/COP ENGINEERING DATA CENTER	16.97	33.60
3.2	MANAGEMENT	0	20.33
4.2	SE&I	0	29.65
5.2	PRODUCT ASSURANCE	0	1.865
6.2	SYSTEM TEST & VERIF	0	54.77
	TOTALS (\$M)	160.05	348.96

Figure 3.6.2-1. Data Handling Facility

DM SBM	WBS MD. DESCRIPTION FY 1987 FY 1990 FY 1991 FY 1992 FY 1994	FY 1987		FY 1988	88			FY 1989	<u>e</u>			FY 1990			Ξ	FY 1991			FY 1992	192			FY 1993	22		=	FY 1994			
	34 40 10 20 30 40 10 20 30 40 10 20 30 40 10 20 30 40 10 20 30 40 10 20 30 40	30 40 10 20 30	9	20	8	3	=	10 20 30 40	2	9	10 20 30	ž.	3	=	25	10 20 39	:	=	10 20 30	3	. #	=	10 20 30	8	9	10 20 30 49	25	7	_	· ·
	MILESTONES	+ CS0																	=	+ 1st Launch	_		+ 100,	+ 10C, POP1 + POP2	P0P2	83 +	å		.	
2.1.1	HARDWARE -NON-RECURRING -RECURRING	0.5 2 1.5 1 1.41 4.41 4.41 2.94 2.94	1.5	1.94	=	3	3	2 46 2	2																				26.4	
2.1.2	SOF THARE	-	.52	1.52	1.52	1.52	1.52	. 52	.52 1.	52 1.	52 1.5	52. 1.5	2 1.5	1.52 1.52 1.52 1.52 1.52 1.52 1.52 1.52	1.52	1.52	1.52	1.52						•					38	_
2.1.3	FAC. INTEGR'N			_	0.22 (0.33	0.22 0.33 0.33 0.33 0.33 0.22 0.22 0.22	.33	.a.	2 6.	22 0.1	æ																	2.2	
TOTALS	TOTALS BY FY QUARTER 1.97 4.94 5.96 5.46 6.15 6.26 6.26 4.79 4.79 1.74 1.74 1.72 1.52 1.52 1.52 1.52 1.52 0 0 0 0 0 0	1.97 4.94 5.96 5.46 6.15 6.26 4.79 4.79 1.74 1.74 1.74 1.52 1.52 1.52 1.52 1.52 1.52 0 0 0 0 0 0 0 0 0 62.6	8	.46	1.15	1.26	6.26	.79	79 1.	7 1	7.1 M	4 1.5	2 1.5	1.52	1.52	1.52	1.52	1.52	•	•	•	. •	•	0	•	•			62.6	•

Figure 3.6.2-2. Data Handler Center

MEI - (Similar To)	YTITI WUD	PER UNIT (SK)	(OR) (NH JATOT I	DOTRE (SK)	ISAN DEVELOPIETT (GK)
Proc. (YAX 8500 Cluster)	3	300	2400		9600
Gatzuay (Data Dist. Not)	. 1	5000	5000	1000	5400
Rationaly (Front End TL®)	1	10000	10000	4000	19999
Tass Stor. (L. O. Hag.T)	4	3000	12000		; ; ; ; ; ; ; ; ; ; ; ; ; ; ; ; ; ; ;
į		i 1 1			!
i		† ! !			
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!		i . ! !			; ; ; ; ; ; ; ; ; ; ; ; ; ; ; ; ; ; ;
, ,		1 ! !	 		
 		1 	! !		i i
TOTALS: {		!	29400	5000	1 25000 1

Figure 3.6.2-3. Level Zero Processing Facility -- GSFC Low Rate

WBS NO.	WBS NO. DESCRIPTION FY 1987 FY 1989 FY 1990 FY 1991 FY 1992 FY 1993 FY 1994	FY 1987		FY 1988	886			F	FY 1989			Ξ	FY 1990			FY 1991	16.			FY 1992	42			FY 1993	5			FY 1994	.	:	
1	39 46 16 29 39 49	30 40 10 20 32	=	10 20 39	æ	9		22	10 20 30	9	=	10 20 30	23	2	2	10 20 30	я	2	=	10 20 30	я	a	9	19 28 39 40 10 20 30 40 10 20 30 40 10 20 30 40 10 20 30 40 10 20 30 40	g	9	9	10 20 30 40	æ	9	18 20 30 40 10 20 30 40 10 20 30 49 10 20 30 40 10 20 30 40 10 20 30 40 10 10 10 10 10 10 10 10 10 10 10 10 10
	MILESTONES	+ CSD																		=	+ 1st Launch	_		+ IOC, POP1 + POP2 + CGP	P0P.	+ 1002	~	8 5 +			
2.2.1	MARDWARE -NOM-RECURRING -RECURRING	0.5 1 1 0.50 0.57 1.14 1.14 1.71 1.71	- Ξ	=	1.71	1.71		=	1.71 1.14 1.14																						2.5
2.2.2	SOFTWARE		0.40	0.40 0.40 0.40 0.40	0.40	9.40		0.40	0.40	0.40	0.40	0.40	9.	0.40	0.40	0.46 0.46 0.40 0.40 0.40 0.40 0.40 0.40	0.40	9.40	0.40	9.40	0.40	0.40	0.40 0	9.40							•
2.2.3	2.2.3 FAC. INTEGR'N			0.08	0.12	0.08 0.12 0.12		0.12	90.0	0.12 0.12 0.08 0.08 0.08	90.0																				9.816
TOTALS	TOTALS BY FY QUARTER 1.07 2.14 2.54 1.62 2.23 2.23 2.23 1.66 1.62 0.48 0.49 0.40 0.40 0.40 0.40 0.40 0.40 0.40	1.07 2.14 2.54 1.62 2.23 2.23	2.54	1.62	2.23	2.23	2.23	1.66	1.62	0.48	0.48	0.40	0.40	0.40	0.40	2.23 1.66 1.62 0.48 0.48 0.40 0.40 0.40 0.40 0.40 0.40	0.40	0.49	0.40	0.40	0.40	0+.4	0.40	3.							23.71

Figure 3.6.2-4. LZPF (GSFC Low Rate)

ITE1 - (Similar To)	YTTRAKLE	PER UNIT (3K)	TOTAL HAY (90)	OOTJE (JK)	IS/// DEVELOPIE IT (3K)
Proc. (MAX %60) Cluster)	4	! 800	3200		i man
Gataway (to Network)	1	5000	5000	1000	1
UNI (Hyperchannel)	1	200	200		
lass Store (Eras. Disk)	1	3000	3000	1500	
-			i ! ! !		
; ;					; ; ;
· }					
!	!				
!					
1			<u> </u>		!
TOTALS: !			11400	2500	1 9999 1

Figure 3.6.2-5. Level Zero Processing Facility -- GSFC High Rate

MBS NO.	MBS ND. DESCRIPTION	FY 1987		FY 1988	96			FY 1989	68			FY 1990	2			FY 1991	_		•	FY 1992			Ę	FY 1993			FY 1994	994		
	30 46 16 20 30	30 40 10 20 30	2	10 20 30	23	a	=	10 20 30	я	=	9	22	10 20 30	10 18 20 30 48 10 20 30 40 10 20 30 40 10 20 30 40 10 20 30 40 10 20 30 40 107ALS	<u> </u>	10 20 30	8	9	9	10 20 30		9	83	10 28 30	2	=	10 20 30 40	9	2	101ALS
	MILESTONES	+ CSD																	•	+ 1st Launch	en Ch		•	10C,PO	+ IBC, POP1 + POP2	0 6 5	4 00	94		
2.3.1	HARDWARE -NDM-RECURING -RECURING	0.5 0.75 1.5 1.25 0.5 1.17 2.34 2.34 2.34 3.51	1.5	2.34	3.51	3.51	3.51	0.5 3.51 3.51 2.34 2.34	2.34																					23.4
2.3.2	2.3.2 SOFTWARE		9.40	0.40 0.40 0.40	0.40	o. to	0,40	0.40	0.40	0.40	9.40	0.40	0.40	0.40 0.40 0.40 0.40 0.40 0.40 0.40 0.40	÷.	9	÷.	0	9		• •	9.	0.4 .4							
2.3.3	2.3.3 FAC. INTEGR'N			0.15	0.22	0.22	0.22	0.22 0.22 0.22 0.15 0.15 0.15	0.15	0.15	0.15																			1.5
TOTALS	TOTALS BY FY GUARTER 1.67 3.09 4.24 4.14 4.64 4.64 4.64 4.14 2.97 2.89 0.55 0.40 0.40 0.40 0.40 0.40 0.40 0.40	1.67 3.09 4.24 4.14 4.64 4.14 2.97 2.89 0.35 0.40 0.40 0.40 0.40 0.40 0.40 0.40 0.4	1.24	= -	3	39.	=	2.97	2.89	35.0	.55 (9	9.4	0 04.0	9	9	9	9	9	0.4	9.0	9	9.0							9

Figure 3.6.2-6. LZPF (GSFC High Rate)

ITE: - (Similar To)	QUANTITY	PER UNIT (3K)	TOTAL (MV) (MK)	DOTES (SK)	(SC) THE ROLLEVED INSI
1		1	3000		
Proc. (VAX 860) Cluster)	4) 300	! 3200 ! ! !		9000
Proc. (Cyber, CRAY)	1	15000	15000	5000	
Rateway (to Network)	1	2000	2000		
UV! (Hyperchannel)	1	200	200		
Tass Storage (Eras. Disk)	1	3000	3000		
į			! !		
i !		1	i i ! !		
;		!	!		
į		į	i i		
; !			! ! ! !		
<u> </u>		<u> </u>	i i		i i
TOTALS: !			; 23400 l	5000	9909

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Figure 3.6.2-7. Level Zero Processing Facility -- JPL High Rate

MBS NO.	MBS NO. DESCRIPTION	FY 1987		7	FY 1988		FY 1989 FY 1990 FY 1991 FY 1992 FY 1993 FY 1994	FY 1989	389			FY 1990	340			FY 1991	=			FY 1992	2		_	FY 1993			•	FY 1994	_	!
		30 40 10 20 30	2	20	8	3	2	19 29 39	贸	\$	=	8	19 29 39 40 19 29 30 40 19 20 30 40	. 9	•	10 20 30	8	2	9	19 20 30 48	8	=	=	19 20 30 40	2	· -	10 20 39 40		9	_
		0SD +				`												• • •		+ ist Launch	Launch			+ 10C,P0P1 + P0P2	190	P0P2	•	a		=
2.4.1	HARDWARE - Non-Recurring - Recurring	0.42 0.84 0.84 0.84 1.26 1.26	0.84	9.94	1.28	1.26	1.26	1.26 0.84 0.84	9. 2																					
2.4.2	SOFTWARE		0.40	6	0.40 0.40 0.40 0.40		9.40	0.40	0.40	0.40	9.49	9.40	0.40 0.40 0.40 0.40 0.40 0.40 0.40 0.40	0.40	0, 40	÷.	9	9.	9	9	9	.6	9	\$						
2.4.3	FAC. INTEGR'N			0.07	0.07 0.10 0.10	0.10	0.10	9.10	0.01	0.10 0.10 0.07 0.07 0.07	0.07																			
		,																												•
TOTALS &	TOTALS BY FY QUARTER	0.42 0.84 1.24 1.31 1.76 1.76	1.24	 	1.76		1.7	1.34	1.31	9.47	0.47	. 8	1.74 1.34 1.31 0.47 0.47 0.40 0.40 0.40 0.40 0.40 0.40	9.40	3.40	9 .	9	9	9	9	9	9	9	9	-	-	-	-	•	<

Figure 3.6.2-8. LZPF (JPL High Rate)

ITE: - (Similar To)	YTITVALE	PER UNIT (SK)	TOTAL HAW (\$K)	DOT&E (SK)	IS/V DEVELOPET (CK)
Proc. (VAX 8600 Cluster)	4	800	3.200		9000
Gataway (to Metwork)	1	2000	2000		i
LN! (Hyperchannel)	1	200	200	;	
lass Storage (Eras. Disk)	1	3000	3000		i
				! !	i
		-			
	}				
TOTALS:		•	l 8400 l		9000

Figure 3.6.2-9. Ground Service Center

	MBS MD. DESCRIPTION	FY 1987		FY 1988	_	į	Ξ	FY 1989	,		FY 1990	940			FY 1991	_		Ξ	FY 1992			FY 1993	993		-	FY 1994			
ļ		30 40	10 20 30	20	0	=	20	10 20 30 40	\$	2	30	10 20 30	\$	9	10 20 30	9	; <u>=</u>	19 29 30	25	.	2	10 20 30	19 20 30 40 10 20 30 40 10	; =	2	07 02 01	07 02		
																										•	•	(8)	
	MILESTONES	• CS0																٠	• 1st Launch	4		•	+ INF PABL + BAB	t Bries	•	ş			
2.5.1	HARDWARE																						5	7.6.	•				
	-MON-RECURRING -RECURRING	0.01	9.28 0.	28 0.4	9 5		900	0. 0. 0. 0. 10. 0.																				0.017	
					:		7																					2.88	
7.6.7	SUPTRAKE	28 0.28 0.28 0.28 0.28 0.28	0.28	28 0.2	2B 0.2E		0.28	0.28	0.28	0.28	0.28	0.28	.28 0	.28 0.	0.28 0.28 0.28 0.28 0.28 0.28 0.28 0.28	28 0.2	78 0.2	9 0.2	•									2	
2.5.3	FAC. INTEGR'H		•	0.03 0.05 0.05	5 0.05		0.02	0.03	0.05 0.05 0.03 0.03 0.03	0.03												·						7.7	
																												0.338	
STELLE	ILIALS BY FY QUARTER	0.42 0.58 0.56 0.60 0.76 0.76	.56 0.	60 0.7	6 0.76	6 0.76	0.61	0.60	0.31	0.31	0.28	7.28 0	. 28 0	.38	0.74 0.61 0.60 0.31 0.31 0.28 0.28 0.28 0.28 0.28 0.28 0.28 0.28	28 0.2	9.2	9 0.28	_									B. 82	

Figure 3.6.2-10. Ground Service Center

ITE: - (Similar To)	CHAMILITY	PS WIT (%)	TOTAL HVW (SK)	DOTCE (SK)	IS/// DEVELOP TENT (SK)
Proc. (W/X 8500 Cluster)	3	550	1650		4(5)
LISP Proc. (Symb. 3670)	1	150	160		540
Cateways (to an. Element)	6	50	300	10	
LN! (Ethernet)	1	15	15		
Mass Storage (DEC RAMI)	20	37	740		
Network Console	1	8	8	2	
	†				i
				; ;	
<u> </u>				i 	
TOTALS:	!	·	2373	12 !	5500

Figure 3.6.2-11. Space Station Operational Control Center

MBS NO	MBS NO. DESCRIPTION	FY 1987		FY 1988	986			FY 1989	384			FY 1990	96			FY 1991	_		٠.	FY 1992 FY 1993 FY 1994			Ξ	FY 1993			Ξ	FY 1994		
	30 40 10 20 30 40	0 . 000	9	10 20 30	20 30	9		20 30	10 20 30	9	9	10 20 39	8	=	=	10 20 30	a	9	10 20 30		9	1 23	8	10 20 30 40 10 20 30 40 10 20 30 40 10 20 30 40 10 26 30 40 10 20 30 40	3	=	æ	10 20 30 40	\$	20 30 40 10 20 30 40 10 20 30 40 10 20 30 40 10 20 30 40 10145 (\$K)
	MILESTONES	6 53 +																	•	+ 1st Launch	unch		-	+ 10C, POP1 + POP2	+	245	2 55 +	25		
2.6.1	HARDWARE -NON-RECURRING -RECURRING	0.21 0.21 0.21 0.42 0.42 0.42 0.42 1.39 2.77 2.77 4.16 4.16	6.21 2.77	0.42 2.77	0.42 4.16	0.42 4.16	0.21 0.21 0.21 4.16 2.77 2.77	0.21 2.77	0.21 2.77																					2.5
2.6.2	SOF THARE		2.9	2.9 2.9 2.9 2.9	2.9	2.9	2.9	2.9	2.9	2.9	2.9	2.9	2.9	2.9	2.9 2.9 2.9 2.9 2.9 2.9 2.9 2.9 2.9 2.9	2.9.	.9	.9	₽.											49.3
2.6.3	FAC. INTESR'N				0.62	0.62 0.62	0.62 0.62 0.62	0.62	0.62																					3.08
TOTALS	TOTALS BY FY DUARTER 1.6 2.98 5.09 8.1 8.1 7.89 6.5 6.5 2.9 2.9 2.9 2.9 2.9 2.9 2.9 2.9 2.9 0.0 0 0 0 0 0 0 0 0 0 82.6	1.6 2.98 5.88 6.09 8.1 8.1 7.89 6.5 6.5 2.9 2.9 2.9 2.9 2.9 2.9 2.9 2.9 2.9 0.0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0	5.88	6.04			7.89	6.5	6.5	2.9	5.2	2.9	2.9	2.9	2.9	2.9 2	.9 2	. 6.	•		0			-				•	•	

Figure 3.6.2-12. Space Station Operational Control Center

ITE: - (Similar To)	YTTIMILE	PER UNIT (SK)	TOTAL HVV: (SK)	DOT&E (3K)	(CD) THE HOLEVER LINE
Processor (IS13088)	3	! 2500		,	 ??500
LISP Proc. (Symb. 3670)	7	160	11.20		44F0
100 Bridge	2	50	100		300
JSC Gataway	1	19000	10000	2000	5000
Core Gateviay	2	30	150	100	420
LANS, Morksta. Connect.	57	30	171C	100	i
LAN, Chinframe Con.	6	45	270	100	! !
Library (DS.I3 8800/3380)	3	456	1368		! ! !
Morksta. (Masscomp MOS50)	55	100	5500	500	16500
TOTALS:		1	27728	2000	49260

Figure 3.6.2-13. Space Station Engineering Data Center

NBS NO.	MBS NO. DESCRIPTION	FY 1987		Ξ	FY 1988			F	FY 1989			Ē	FY 1990			FY 1991	166			FY 1992	2		_	FY 1993			Ŧ	FY 1994	
	30 46 16 28 39 48	30 44 16 28 30		16 28 39	Ä	#		8	10 20 30	9	2	2	10 20 30 40 10 20 30 40 10 20 30 40 10 20 30 40 10 20 32 40 10 20 32 40	2	9	10 20 30	10 20 30	3	=	10 20 30 40	8	9	•	10 20 30	19 28 30 40 19 26 30 40	=	92	10 26 30 40	TOTALS (8M)
	MILESTONES	+ CS0																		÷ :=	+ 1st Launch		-	, 10C, P	+ 10C, P0P1 + P0P2	P0P2	+ 000	<u>&</u>	
2.7.1	HARDWARE -NOW-RECURRING -Recurring	0.1 0.1 0.1 0.2 0.2 0.2 0.85 1.7 1.7 1.7 2.55 2.55	9		2 0. 7 2.5	2 6.2	2 0.1 5 2.55	1.7	1.7.1																				1.2
2.7.2	SOFTWARE		<u>:</u>	1.39 1.39 1.39 1.39	9 1.3	9 1.3		1.39	1.39	1.39	1.39	1.39	1.39 1.39 1.39 1.39 1.39 1.39 1.39 1.39	1.39	1.39	1.39	1.39	1.39	1.39	1.39	£.	. es.	.39	33					
2.7.3	FAC. INTEGR'N				6.3	0.38 0.38		9.38	0.38 0.38 0.38																				
TALS BY	TOTALS BY FY DUARTER 0.95 L.8 3.19 3.29 4.52 4.42 3.57 3.57 1.39 1.39 1.39 1.39 1.39 1.39 1.39 1.39	0.95 1.8 3.19 3.29 4.52 4.52		19 3.25	4.5	2 4.5	7.42	3.57	3.57	87	1.39	2	87	2	2	82	9	9	9	27 1 27 1 37 1 37 1 37 1 37 1 37 1 38 1 38 1 3	9	9	2	9					£ 95

Figure 3.6.2-14. Space Station Engineering Data Center

TE:-(Similar To)	CURVITITY	PER UNIT (SK)	! TOTAL HVY (9K)!	DOTUE (SK)	ISAN DEVELOP ENT (CK)
Processor (1743098)	4	2500	10000		30000
Gateway (to JSC Met.)	2	30	160	100	450
LAN, Mainframe Con.	8	45	360	100	
iag. Disk (IS: 3880/3380)	1	456	456		1
Optical Disk	1	6000	5000	1000	
i !			 		
!			i i		
i !			; ! !		
!	:		i 		
i !	 	i 	i ·		
TOTALS: 1		 	16976	1200	30480

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Figure 3.6.2-15. Platform Control Center

MBS NO.	MBS NO. DESCRIPTION FY 1987 FY 1988 FY 1989 FY 1990 FY 1991 FY 1993 FY 1994	FY 1987	i	=	FY 1988	;		FY 1989	FY 1989	;		Ŧ	FY 1990			F	FY 1991			FY 1992	266			FY 1993	993			FY 1994	_	
	30 40 10 20 30 40 11 20 30 40 10 20 30 40 10 20 30 40 10 20 30 40 10 20 30 40	39 40 10 29 39	9	78	25	2	9	20	8	2	=	23	28 38 46 17 20 30 46 19 20 36 46	2	2	10 20 30	я	, =	.9	10 20 30 40	83	, 2	9	10 20 30 40	25	. 9	2	10 20 30 40	9	40 TOTALS
	MILESTONES	050 +																		+	+ Ist Launch	€		÷	1,404,2	+ 10C, POP1 + P0P2	2	40J +		:
2.8.1	HARDWARE -NOM-RECURING -RECURING	0.23 0.23 0.23 0.46 0.46 1.15 2.29 2.29 2.29 3.44	5 0.23	2.29	3.4	3.5	5.54	0.23 0.23 0.23 3.44 2.29 2.29	0.23																					
2.8.2	SOFTWARE		1.63	1.63	1.63 1.63 1.63 1.63 1.63 1.63 1.63 1.63	1.63	3 1.63	1.63	1.63	1.63	1.63	1.63	1.63	1.63	1.63	1.63	1.63	1.63	1.63	1.63	1.63	1.63	1.63	1.63	1.63					
2.8.3	FAC. INTEGR'N				0.66	9.66		0.66 0.66 0.66	0.66																					
TOTALS 1	TOTALS BY FY QUARTER 1.38 2.52 4.15 4.39 6.2 6.	1.39 2.52 4.15 4.39 6.2 6.2	4.15	3.4	4.2		***	5. 4. 81 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1	4	17	17.			5		:			1									2 60. 40. 40. 40. 12		:

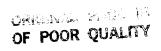


Figure 3.6.2-16 . Platform Control Center

ITE: - (Sinilar To)	CUANTITY	PER UNIT (SK)	TOTAL HVY (9K)	00T&E (GK)	ISA: DEVELOPET (CK)
Processor (IS13083)	3	2500			22500
1	J	!			1
LISP Proc. (Symb. 3670)	5	150 !	300		3200
GSFC Gattoway	1	10000	10000	2000	5700
Core Gateway	2	ဆ	160	100	400
LV's, Morksta. Con.	22	30	560	100	, , , , , , , , , , , , , , , , , , ,
UNU lainframe Con.	ć	45	270	100	; ; ; ; ; ; ; ; ; ; ; ; ; ; ; ; ; ; ;
Library (DS,IR 8080/3380)	3	455	1366		1 1 2 2 1 1
Monksta. (Masscomp 10550)	21	100	2100	500	5300
		r 	! !		† †
		 			! !
TOTALS			1 22858 1	2300	1 37490 1

Figure 3.6.2-17. POP/COP Engineering Data Center

#BS NO.	MBS NO. DESCRIPTION	FY 1987			FY 1988	2			FY 1989	86			FY 1990	940			FY 1991	161			FY 1992	33			FY 1993	FY 1989 FY 1990 FY 1991 FY 1992 FY 1994		ш.	FY 1994	
	30 40 18 20 30 40 16 20 30 40 10 20 30 40 10 20 30 40 10 20 30 40 10 20 30 40	30 40 10 20 30		10 20 32	39	25	=	=	10 20 30	20 30 40 10 20 30	2	2	æ	10 20 30	.	•	8 7	10 20 30 40	#	2	10 20 30 40	8	2	2	10 20 30	10 20 30 40 10 20 30 40	9	7	10 20 30 40	D TOTALS
	NILESTONES	+ CSD																			+	+ 1st Launch	_		, 10C,	+ 10C, P0P1 + P0P2	P0P2	•	an O	ı
2.9.1	MARDWARE - NDM-RECURTING - RECURKING	0.1 0.1 0.1 0.2 0.2 0.2 0.1 0.1 0.1 0.1 0.1 0.1 0.1 0.1 0.1 0.1	1.7	0.1	0.2	0.2	0.2 7.55 2	 	0.1	1.7																				
2.9.2	SOFTWARE		-	.39	.39	.39	. 39	. y	85	39	ñ	1.39	1.39	1.39	1.39	1.39	1.39	1.39	65.1	. 39	1.39 1.39 1.39 1.39 1.39 1.39 1.39 1.39	. 35	. 39	8	55					
2.9.3	FAC. INTEGR'N				-	0.38 0.38		0.38 0.38 0.38	8 7.	8.			i																	
TOTALS BY	TOTALS BY FY BUMBRIER 0.95 1.8 3.19 3.29 4.52 4.52 4.42 3.57 3.57 1.39 1.39 1.39 1.39 1.39 1.39 1.39 1.39	0.75 1.8 3.19 3.29 4.52 4.42 3.57 3.57 1.39 1.39 1.39 1.39 1.39 1.39 1.39 1.39		2 61	8	25	22				2	2	2		2															

Figure 3.6.2-18. POP/COP Engineering Data Center

ITE - (Similar To)	YTTT! PAUC	FER UNIT (3K)	TOTAL HAY (SK)	DOTRE (SK)	SAT DEVELOPIENT (SK)
Processor (I313093)	4	2500	10000		30000
Gaterray (to GSFC)	2	30	160	100	490
LAW lainframe I/F	8	45	360	100	
 Hag Disks (1911-3880/3380	1	456	456		
Optical Disk Unit	1	5000	6000	1990	
					! !
		i 			
i !		· .	1	1	
i 		! !	! !		!
 		! !	! !		
TOTALS: !		!	16976	1200	30480

Figure 3.6.2-19. Management

HBS NO.	DESCRIPTION	FY 1987	187		FY 1988	28			FY 1989	2			FY 1990	Y 1990	,	-	FY 1991			_	FY 1992			٤	FY 1993			FY 1994	994		
		2	2	=	22	e	₽.	=	20	3		<u>=</u>	22		-	9	10 20 34		9	10 2	10 20 30		9	92 8	8	2	9	8	23	9	TOTALS (8N)
*	MILESTONES	+ CSD	_																	•	+ 1st Launch	aunch		•	+ 10C,P0P1 + P0P2	÷	240	a 00 •	9-		
3.2.1	DATA MANDLING CENTER	0.15	0.15 0.15 0.15 0.15 0.15 0.15). 15	0.15	0.15		0.15 0.15		0.15 0	0.15 0	. 15	.15	.15	.15	15 0.	0.15 0.15 0.15 0.15 0.15 0.15 0.15	15 0.	0.15 0.15		0.15										m
3.2.2	LZPF - 65FC LOW RATE	0.04 0.04 0.04 0.04 0.04 0.04	0.04).04	9.0	0.04		0.04	2.0	20.0	3	5	5	20.	9	2	0.04 0.04 0.04 0.04 0.04 0.04 0.04 0.04	94 0.0	0.04 0.04		0.04 0.04 0.04 0.04	3	34 P.C	¥ 0.0	,						Ξ
3.2.3	LIPF - 6SFC HIGH RAIE	0.08 0.08 0.08 0.08	90.0	90.(90.0	8.0	0.08	90.0	0.08	0.08	0.08	90.0	0.08 0.	0.08 0.	0.08 0.	0.08 0.	0.08 0.08	90.08	9	0.08	0.08 0.08	80 · 0 80	90.08	90.08	90						2.05
3.2.4	LIPF - JPL HIGH RATE 0.04 0.04 0.04 0.04 0.04 0.04	9.0	0.04	.04	9.04	0.04		.04	20.0	20.0	0.04 0.04 0.04 0.04 0.04 0.04 0.04 0.04	5	2.	2.	9	2	0.04	94 0.04	<u>.</u>	0.04 0.0	0.04 0.04	94 0.04	0.0 0.04	9.0	-						0.957
3.2.5	GROUND SERVICE CENTER 0.02 0.02 0.02 0.02 0.02	0.05	0.02 0	.02 (0.02 (D. 02 C		0.02	0.02	0.02	0.02 0.	0.02 0.	0.02 0.	0.02 0.	0.02 0.02	02 0.	0.02 0.03	02 0.02	02 0.02	02 0.02	05										0.465
3.2.6	SS DP. CONTROL CENTER 0.22 0.22 0.22 0.22 0.22 0.22	0.22	0.22 0	.22.	0.22 (D. 22 C		0.22 0.22		.22	0.22 0.22 0.22 0.22	.22 0.	.22 0.	0.22 0	0.2 0.22 0.22	22 0.	22 0.22	22 0.22	22 0.22	22											4.24
3.2.7	SS ENGR. DATA CENTER	0.10 0.10 0.10 0.10 0.10 0.10	0.10	0.10	9.10). 10 G		0.10	0.10	0.10 0.10	9	6.10 0.	9	0.10 0.1 0.10 0.10 0.10	50 0.	9	6.1	9.7	0		0.10 0.10 0.10 0.10 0.10 0.10 0.10	0.1	0.1	0 0.10	•						2.6
3.2.8	P/F CONTROL CENTER	0.13 0.13 0.13 0.13 0.13 0.13	0.13 0	1.13	9.13 ().13 G		0.13 0.13		0.13 0.13	.13 0.	0.13 0.	0.13 0.	0.13 0.13 0.13	13 0.	13 0.	0.13 0.13	3 0.1	0.13 0.13	13 0.13	13 0.13	13 0.13	13 0.13	3 0.13	3 0.13	_					3.32
3.2.9	PDP/CDP EMS. DATA CTR. 0.10 0.10 0.10 0.10 0.10 0.10	0.10	0.10	.10).10 C	0.10		0.10 0.10		0.10	0.10 0.	0.10 0.	0.10	0.1 0.10 0.10	9	10 0.	0.10 0.10 0.10 0.10 0.10 0.10	0 0.1	9.1	10 0.1	0.	0.10	0.10	0 0.10							2.6
4 4 44 44	TATAL BO TO DISCOURT						- 1																					:			
	LUTALS BY THE BEARIES		0.91 0.91 0.91 0.91 0.91 0.91	.9 16:	. 91	- -		3 F.	ē.	ē.	ج و	٠ و		ە 2		91	£ 0.3	5.0	9.	3. 1	59 0.:	2.0.5	12 0.5	2 0.5	0.91 0.91 0.91 0.91 0.91 0.91 0.90 0.89 0.91 0.91 0.91 0.91 0.91 0.90 0.52 0.52 0.52 0.52 0.13	•	•	•	•		20.33

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Figure 3.6.2-20. System Engineering and Integration

MBS NO.	MBS ND. DESCRIPTION FY 1987 FY 1988 FY 1989	Ξ	FY 1987		Ξ	FY 1988			-	FY 1989			FY 1990	FY 1990			Œ	FY 1991			1	FY 1992			Œ	FY 1991 FY 1992 FY 1993 FY 1994			€	FY 1994			
		æ	2	=	5 0		#	9	10 20	g 9	2	i	10 20	:	∓ 83	9	10	20 30	#	2	1	55 82	95 95	2	70	8	9	=	92	8	7		TOTALS (SN)
	MILESTONES	6 SJ +	es :																		•	+ ist Launch	inec p		•	+ 10C, POP1 + POP2	÷	P0P2	•	80 3 +			
4.2.1	DATA HANDLING CENTER		• o. •	•	0.44 0.44 0.44 0.44 0.26	9	26 0.26	_	0.26 0.22	22 0.17	.0 (1	13 0.1	13 0.	13 0.	13 0.	0.13 0.13 0.13 0.13 0.13 0.13 0.13 0.13		3 0.1	3 0.1	3 0.1	13 0.13	22										-	:
4.2.2	LZPF - 6SFC LOW RATE		9.1	6 0.1	0.16 0.16 0.15 0.16 0.08	16 0.0	90.08	_	0.08 0.08	08 0.08	80.0 80	90.0 80	0.03	03 0.1	0.03 0.0	0.03 0.03 0.03 0.03 0.03 0.03 0.03 0.03	33 9.(3 0.0	3.0.0	3.0.0	3 0.0	3 0.1	3. 0.6	3 0.03	3 0.03	2							1.6
4.2.3	LIPF - 6SFC HIGH RATE 0.29 0.29 0.29 0.29 0.15 0.15	TE 0.29	9 0.2	9 0.2	9 0.2	6.	15 0.	_	55 0.	15 0.3	5	15 0.1	12 0.(96 0.1	96 0.1	0.15 0.15 0.15 0.15 0.12 0.06 0.06 0.06 0.06 0.06 0.06 0.06 0.0	90.0	0.0	9.0	9	9.0	96 0.1	9.0	90.00	90.00	.							2.99
4.2.4	LZPF - JPL HISH RATE		- 6	0.14 0.14 0.14	1.0	0.14 0.07	07 0.07	_	0.0 0.03	0.0 0	37 0.(0.07 0.05	05 0.02	0.0	0.02 0.0	0.02 0.0	2 0.1	0.02 0.02 0.02 0.02	2 0.6	20.02	0.02	20.0 20	0.03	2 0.02	2 0.02	~						-	:
4.2.5	BROUMB SERVICE CENTER 0.06 0.06 0.06 0.06 0.06	ER 0.06	5 0.0	0.0	9.0	96 0.4	04 0.04	•	9.	0.04 0.03 0.02	2 0.0	0.02 0.02	02 0.02	02 0.02	02 0.1	0.02 0.02 0.02 0.02 0.02	2 0.4	0.0	2 0.6	2 0.03	0.02	2										9.678	82
4.2.6	SS DP. CONTROL CENTER 0.51 0.51 0.51 0.51 0.51	ER 0.51	1 0.5	1 0.5	ii 0.5	31 0.1	51 0.51	_	31 6.	31 0.	2 0.	H 0.2	Z 0.2	21 0.	21 0.5	9.31 6.31 6.31 6.31 6.21 6.21 6.21 6.21 6.21 6.21 6.21 6.2	31 0.	1 0.2	- 0.7	21 0.21	z :											6.16	=
4.2.7	SS ENGR. DATA CENTER	R 0.31	0.3	1 0.3	6.31 0.31 0.31 0.31 0.31		31 0.31		0.12 0.12	12 0.12	12 0.12	12 0.12	12 0.12	12 0.09	9 0.09	99 0.09	9 0.09	9 0.09	9 0.09	9 0.09	90.00	90.09	99 0.09	9 0.09	9 0.09	9 -						m	3.8
4.2.B	P/F CONTROL CENTER	•	•	•	÷	•	0.4 0.4 0.4 0.4 0.4 0.4	•	9	16 0.1	16 0.	9.1	16 0.1	16 0.		0.16 0.16 0.16 0.16 0.16 0.16 0.11 0.11	1.0.1	1.0	- 0.1	- 9.1	1.0.1	.0	0.11 0.11	1 0.11	1 0.11	- 0. II	_					7	4.83
4.2.9	PUP/EDP ENG. DATA ETR. 0.31 0.31 0.31 0.31 0.31 0.31	TR. 0.31	0.3	1 0.3	1 0.3	-	 	_	12 0.	0.12 0.12 0.12	7 0.1	0.12 0.12	12 0.12	12 0.09		0.09 0.09	9 0.09	9 0.09	9 0.09	9 0.09	9.09	9 0.09	9 0.09	9 0.09	9 0.09	<u>-</u>						ri	3.8
TOTALS	TOTALS BY FY DUMPTER 2.65 2.65 2.65 2.14 2.14	2.65	2.6	3 2.6	5 2.6	5 2.1	2.65 2.65 2.65 2.65 2.14 2.14		1 21	32 1.27 1.22 1.17 1.	7 7	7 1.0	9.0	9	6.0	7.9 6.7	0 0	6.0	6.7	0.0	8.0	9.0	7 0.4	2 0.4	9	0.78 6.78 6.78 6.78 6.78 6.57 6.42 6.42 6.42 6.11			0			. 72 1. 72 1. 12 1. 10 1. 0. 89 0. 78 0. 78 0. 78 0. 78 0. 78 0. 78 0. 57 0.42 0.42 0.42 0.42 0.42 0.11 0. 0 0 0 0 0 0.78 0.78 0.78 0.78 0.78 0	1

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Figure 3.6.2-21. Product Assurance

Sem	MBS NO. DESCRIPTION FY 1987 FY 1988	Ē	FY 1987		ž	FY 1988		Ξ	FY 1989			٠.	FY 1990			FY 1991	166			FY 1992	992			FY 1993	2			144			
		2	30		10 20 30	8	#	2	10 20 30	\$	2	! ~ !	97 98	2	9	10 29 30	23	=	2	20	g	\$	=	22	9	9	2	20 30 40	200		TOTALS (SM)
	MILESTONES	653 +	ş			; ; ;														-	+ Ist Launch	5		<u> </u>	1904	+ 10C,P0P1 + P0P2		3			
5.2.1			0.0	1 0.0	0.01 0.01 0.01 0.01 0.01	0.0	9.0	0.0	0.01	0.01	0.01	9.0	0.01	0.01	0.01 0.01 0.01 0.01 0.01 0.01 0.01 0.01	0.01	0.01	0.01	9.0	0.01										0.277	
5.2.2		TE 0.00	9.0	9.0	0.0	9.	0	0.00 0.00	9.6	8.8	0.00	9.0	9.00	9.0	0.00 0.00 0.00 0.00 0.00 0.00 0.00 0.00	8.8	8.	9.0	8.6	9.8										0.102	2
5.2.3		ATE 0.00	9.0	9.0	0.00 0.00 0.00 0.00	0.0	0.00 0.00	0.00 0.00	9.0	8.6	9.0	9.6	0.00	9.6	0.00 0.00 0.00 0.00 0.00 0.00 0.00 0.00 0.00 0.00	9.0	0.00	9.	8.	8.	8.8	0.00 0.00		8.0						0.197	
5.2.4	4 LIPF - JPL HIGH RATE	TE 0.00	0.0	9.0	0.00 0.00 0.00 0.00 0.00 0.00	0.0	0.0	9.0	9.0	8.	8	0.00 0.00 0.00 0.00 0.00 0.00	8.8	9.0	0.00 0.00 0.00 0.00 0.00	9.0	9.60	8.	0.00 0.00	8.	0.00 0.00	8.0	0.00	0.00						0.087	
5.2.5	5 GRUJUND SERVICE CENTER 0.00 0.00 0.00 0.00 0.00	TER 0.00	9.0	9.0	9.0	0.0	0.0	0.00 0.00	9.0	0.00 0.00		8.	8.	8.8	0.00 0.00 0.00 0.00 0.00 0.00 0.00	9.6	9.8	8.	8.	8.										0.042	45
5.2.6	b \$5 GP. CONTROL CENTER 0.02 0.02 0.02 0.02	TER 0.02	0.0	2 0.0	2 0.0	2 0.0	12 0.0	0.02 0.03	2 0.02	0.02	0.02	0.02	0.05	0.05	0.02 0.02 0.02 0.02 0.02 0.02 0.02 0.02	0.05	0.03	0.03	0.03											0.39	ě
5.2.7	7 SS ENSR. DATA CENTER	ER 0.01	0.0	0.0	0.01 0.01 0.01 0.01 0.01	1 0.0	0.0	0.0	0.0	0.0	0.01	0.01	0.01	0.01	0.01 0.01 0.01 0.01 0.01 0.01 0.01 0.01	0.0	9.0	0.01	0.01	0.01	0.01 0.01 0.01 0.01 0.01	0.01	10.0	0.01						0.24	*
5.2.8	8 P/F CONTROL CENTER		1 0.0	0.0	0.01 0.01 0.01 0.01 0.01 0.01	0.0	0.0	0.01 0.01	1 0.01	10.01	0.0	0.0	0.0	0.01	0.01 0.01 0.01 0.01 0.01 0.02 0.01	0.03	0.01	0.0	0.01	0.01	0.0	9.0	0.0	0.01	0.01					•	0.3
5.2.9	9 POP/COP EMS. DATA CTR. 0.01 0.01 0.01 0.01 0.01	CTR. 0.01	0.0	0.0	0.0	0.0	0.0	0.0	0.01 0.01 0.01	0.0	0.01	0.01	0.0	9.0	0.01 0.01 0.01 0.01 0.01 0.01 0.01 0.01	0.0	0.0	0.0	0.0	0.0	0.01		0.0	0.01						•	0.24

TOTALS BY FY QUARTER

Figure 3.6.2-22. System Test and Verification

ANIESTORES	NBS NO.	NBS NO. DESCRIPTION	FY 1987		Œ	FY 1988	_		Œ	FY 1989			Ξ	FY 1990			FY 1991	166		FY 1992	FY 1992	992			FY 1993	2.		FY 1994	FY 1994		:	
## FIRESTONES FISE				i	2.0	, m	!		1	i	: :	į	1	; i	! }		8	я		9	3 8	87		2	8	8	2	i	į	1	_	S .
DATA HANDLING CENTER 0.33 0.33 0.49 0.49 0.49 0.49 0.49 0.49 0.40 0.83 0.83 0.83 0.83 0.83 0.83 0.83 0.8		NILESTONES	¢ CS0																		*	t Laum	5		÷ 100	P0P.	+ 100		3		<u>.</u>	:
L7PF - 65FC LOW RATE 0.12 0.12 0.12 0.12 0.12 0.12 0.12 0.13 0.18 0.18 0.18 0.18 0.18 0.18 0.24 0.24 0.24 0.24 0.24 0.24 0.24 0.24	6.2.1	DATA HANDLING CENTER					•		33 0.3	3 0.33	- 0.	9 0.49	0.49	0.4	0.49	0.49	9.6	0.B3	0.83	0.63											•	_
L2PF - 58FC HIGH RAIE L2PF - 7PL HIGH RAIE 6.019 0.19 0.19 0.19 0.19 0.19 0.19 0.19	6.2.2	LZPF - 6SFC LOW RATE								2 0.12	2 0.12	0.12	0.12	0.12	9.1	0.18	9.18	9.18	0.18	9. 18	0.24	0.24	0.24	0.24	0.24						.,	_
6ROUND SERVICE CEMTER 6.05 6.05 6.05 6.05 6.05 6.07 6.07 6.07 0.07 0.07 0.07 0.07 0.07	6.2.3	LZPF - 6SFC HIGH RATE							-:	9 0.15	0.15	6.19	0.19	0.19	0.28	0.28	0.28	0.28	0.28	0.28	0.38	0.38	0.38		0.38						7	
SS OP. CONTROL CENTER SS ENGR. DATA CENTER	6.2.4	LZPF - JPL HIGH RATE								0 0.16	0.10	0.10	0.10	0.10	0.15	0.15	0.15	0.15	0.15	0.15	0.20	0.30	0.20		0.20						7.7	
55 OP. CONTROL CENTER 55 OP. CONTROL CENTER 55 EMBR. DATA CENTER 56 EMBR. DATA CENTER 57 EMBR. DATA CENTER 58 EMBR. DATA CENTER 59 EMBR. DATA CENTER 59 EMBR. DATA CENTER 50.27 0.27 0.27 0.27 0.27 0.27 0.37 0.34 0.34 0.34 0.34 0.34 0.35 0.59 0.59 0.59 0.59 50 EMBR. DATA CENTER 50.29 0.29 0.29 0.29 0.29 0.29 0.39 0.35 0.35 0.35 0.35 0.35 0.57 0.57 0.57 0.57 0.57	6.2.5	GROUND SERVICE CENTER					9.		0.0	5 0.05	0.03	0.07	0.0	0.01	0.07	0.0	0.10	0.12	0.12												1.2	_
55 EMBR. DATA CENTER 0.22 0.22 0.22 0.22 0.34 0.34 0.34 0.34 0.34 0.34 0.35 0.59 0.59 0.59 0.59 0.59 0.59 PJF CONTROL CENTER 0.29 0.29 0.29 0.29 0.29 0.29 0.29 0.30 0.34 0.44 0.44 0.44 0.44 0.44 0.64 0.64 0.6	4.2.4	SS OP. CONTROL CENTER						•	\$ 0°.	6 0.44	9.48	0.98	0.98	0.98	0.98	1.16	1.1	1.16	÷:	1.16											=	
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	6.2.9	POP/COP ENG. DATA CTR.						0.3	18 0.2	3 0.28	0.28	0.28	0.35	0.35	0.35	0.35	0.35	0.35	0.35	0.35	0.57	0.57	0.57	0.57	0.57						7.12	
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3.6.2.2 Operational Costs

Operational staffing for each of the ground facilities was estimated in this costing effort. As noted in the ground rules and assumptions, it was difficult to separate generic staffing from mission unique requirements therefore that visibility is not provided. In addition, the staffing is provided based on 24-hour coverage. These operational costs in man-years per year are provided in Table 3.6.2-2.

Table 3.6.2-2

GROUND SEGMENT OPERATIONAL COSTS

FACILITY	STAFFING (MANYEARS/YEAR)
DHC	36
LZPF GSFC LO-RATE	52
LZPF GSFC HI-RATE	52
LZPF JPL HI-RATE	28
GSC	32
SSOCC	276
SSEDC	4
P/F CC	136
POP/COPEDC	36

3.6.3 Total SSDS Implementation Costs

The total implementation costs for the SSDS is simply the summation of the Onboard and Ground Systems costs provided in sections 3.6.1 and 3.6.2. This total is \$1,143.58M and consists of \$339.02M Recurring plus \$804.56M Non-Recurring. Distribution of these costs across the WBS elements is provided in Table 3.6.3-1.

3.6.4 Code S/Code T Estimated Funding Responsibilities

As part of the costing estimation effort, NASA requested estimates of the Code S and Code T element funding responsibilities. The first effort was to develop an allocation of Onboard and Ground System components to a Code S/Code T implementation/operation responsibility matrix. This matrix, provided as Appendix A to this report, was generated based on NASA provided inputs and guidelines.

Based on the allocation matrix, the total element costs are:

Code S: \$814.9M Code T: \$328.7M

Time-phased cost distributions of the allocated costs are provided in Figure 3.6.4-1.

Table 3.6.3-1

DISTRIBUTION OF COSTS ACROSS WBS

Segment	H/W Devel (\$M)	H/W Recur (\$M)	S/W Devel (\$M)	Integ. (\$M)	Mgmt (\$M)	Syst Engr & Integ. (\$M)	Product Assoc. (\$M)	Product Syst Test Assoc. & Verif. (\$M) (\$M)	GSE (\$M)	Total (\$M)
Space	63.2	179.0	178.1	19.0 ו 25.1	82.7	66.2	7.2	12.5	20.6 634.6	634.6
Ground	20.2	160.02	206.39	15.75	60.02 206.39 15.75 20.33	29.65	1.87	54.77	0	0 508.98
TOTALS(\$M) 83.4	83.4	339.02	384.49	39.02 384.49 40.85 103.03	103.03	95.85	9.07	67.27	20.6	20.6 1,143.58

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Figure 3.6.4-1. Time-Phased Cost Distributions '

NASA CODE S

. OH 562	, DESCRIPTION	FY 1987	9		FY 1988	92			FY 1989	\$			FY 1990	9			FY 1991	_		;	FY 1992	~		!	FY 1993	İ			FY 1994			
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	SPACE BESN'T ALLOC'N 14.6 33.2 21.7 21.7 31.1 31.1	7.5	13.2	21.7	21.7	31.1		2 .2	¥.2			12.4	32.4	1.11	32.5	и.7	29 12.4 32.4 9.77 32.5 33.7 33.7 24.1 24.4 24.2	¥.:	4.4 2	4.2 2	24.2 16.1	¥::	=======================================	11	16.1 13.1 12.8 2.12 2.12 1.06 1.06	12 2.1	1.6	 	2	•	9 634.6	•
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101 1 3000	ALS BY FY QUARTER	3.31 4.45 6.00 6.32 7.46 7.99	3.31 4.45 6.08 6.32 7.46 7.99	8.	7	7.1	7.3		3.5	17	1 65	21.	1 22	1.17	8.	8	8	1 5	7 4.	5 5.	32 t.		1 5	17 6	1.07 7.30 7.25 4.19 4.13 4.22 4.17 4.38 4.35 4.71 4.71 4.90 5.32 4.19 4.19 4.19 2.53	-		•			138.7	_

4.0 SSDS Program Schedules

Program scheduling is necessarily the fundamental effort of any successful Program Plan. It provides the time-lines and key project milestones against which the cost and man-power resources must be metered. It allows over-view of critical task phasing in supporting initial feasibility determinations against project milestones and provides an on-going tool to evaluate effects of program perturbations.

4.1 Introduction

The intent of this section is to provide over-view schedules of the SSDS implementation efforts. The schedules are consistent with the Work Breakdown Structure (WBS) of Section 3.3 although scheduling resolution is only to the third level (X.X.X) WBS elements. Separate schedules are provided for the On-Board and Ground Systems primarily to maintain manageable presentations.

4.2 Scope

The schedules provided are limited to the SSDS On-Board and Ground Systems as basically defined in the Task 4 Report, Sections 6.0 and 7.0 respectively. The schedule provided are in the form of simple bar charts. Gantt or Pert chart formats identifying the underlying complexities, critical paths, etc., are beyond the remaining resource capability of the study.

As in the case of the costing effort, the SSE and DSIT elements of the SSDS have not been addressed.

4.3 Ground-Rules and Assumptions

The following represent the significant ground-rules and assumptions utilized in developing the schedules.

1. Major program milestones have been adopted from those of the current Level C/D planning

- 2. The first platform, defined by the Langley Data Base as POP 1 will be deployed coincident with Station IOC; POP 2 and then the COP will follow on 6-month intervals. POP 3, also identified in the Langley Data Base has not been addressed.
- 3. The 'core support' elements of the SSDS Ground System must be in place and operational to support the first launch.

4.4 Methodology

The methodologies utilized for the schedule development have been engineering estimations based on insight and experience with prior programs.

4.5 <u>Schedules</u>

The On-Board system schedule and required WBS defined support item and Ground Systems schedule and required WBS defined support items are provided in Figures 4.5.1 and 4.5.2 respectively.

Figure 4.5-1

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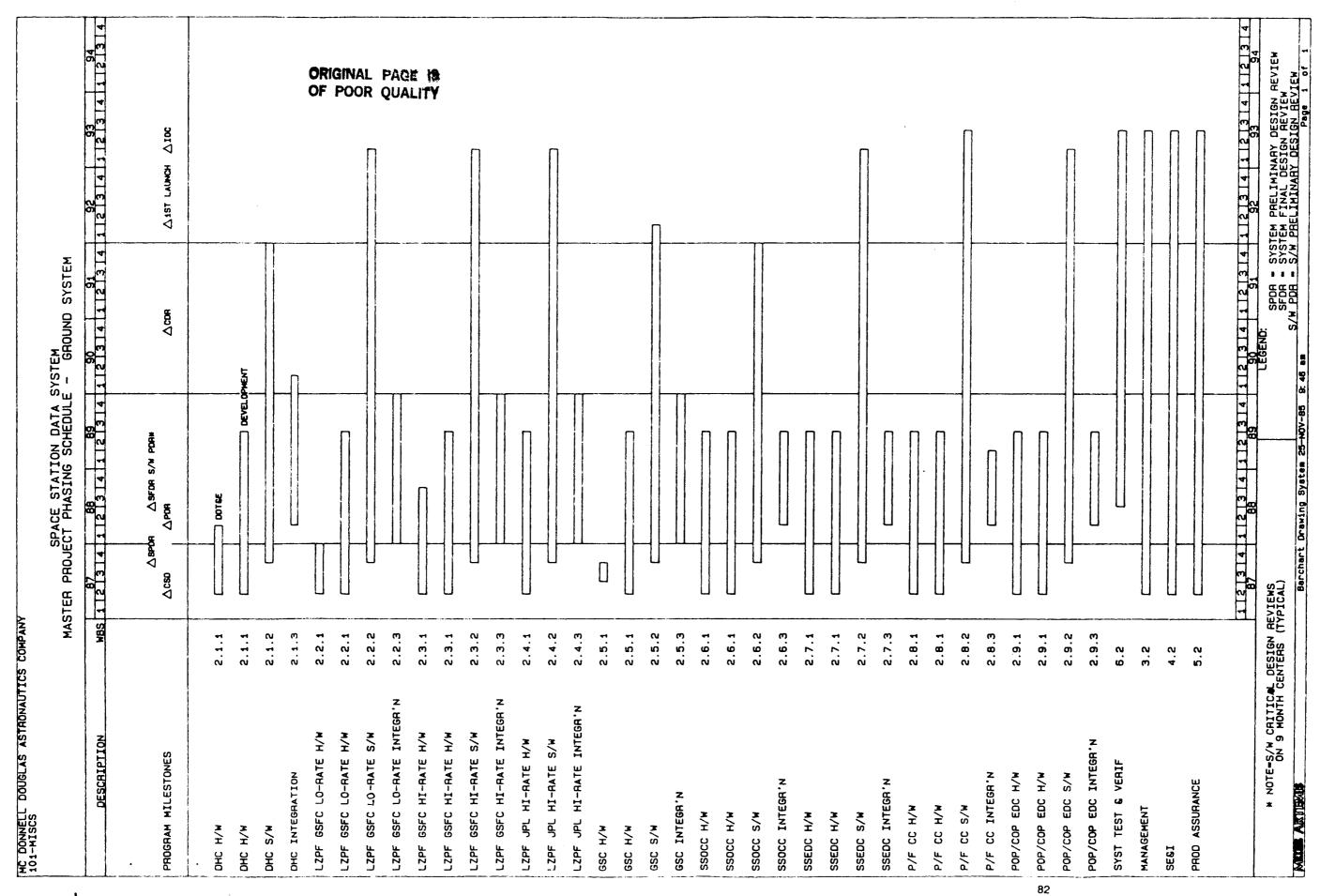


Figure 4.5-2

5.0 ADVANCED TECHNOLOGY DEVELOPMENT RECOMMENDATIONS

5.1 INTRODUCTION

NASA has forecast technology development requirements for the Space Station Program based on earlier studies/analysis of the developing design requirements, and is supporting/funding those technologies through advanced development (A/D) project plans. It is the intent of this section to evaluate technology issues identified by the MDAC SSDS A/A Study against these NASA advanced development plans.

A summary of the NASA SSDS applicable advanced development tasks is provided in Section 5.3. Advanced Technology Development candidates identified by the Study are provided in Section 5.4 and the conclusions/recommendations resulting from the evaluation is provided in Section 5.5.

5.2 SCOPE

This effort will be limited to an assessment of those technologies directly utilized by the MDAC Study Task 4 System Definitions and to technologies that were discounted during the Task 3 trade studies because of "development risk". By NASA direction, only NASA sponsored development activities will be addressed; current/proposed industry IR&D activities have not been considered and were not factored into the Section 5.5 recommendations.

5.3 NASA ADVANCED TECHNOLOGY DEVELOPMENT SUMMARY

The NASA advanced development summary provided in this section is an extraction from two basic sources. The first is the "Space Station Data Management Advanced Development Project Plan" provided directly by NASA; the second source is the Commerce Business Daily which was surveyed for additional NASA sponsored activities in the SSDS technology areas.

The items in this section represent the study team assessment of the over-all NASA SSDS oriented advanced development plans.

- o SSE Development Activities JSC: Development of direct analytical support required for synthesis of SSE requirements in coherent formulations suitable for end-item management products. JSC also to provide a prototyping (SSE Computer) capability for evaluation of SSE software.
- o ADA Evaluation/Implementation JSC, Marshall Space Flight Center (MSFC), National Space Technology Laboratory (NSTL): JSC to develop guidelines for use of Ada for SSE and to evaluate detailed implications of Ada applied to 'real time' applications requiring rigorous verification. NSTL to code the Payload Simulator (Ref. Task 40) in Ada.
- o Network/Bus Interface Units JSC, GSFC: GSFC to design/prototype a hardware BIU for testing the GSFC Star Bus system. JSC to produce a commercially-derived "Multi-compatible Network Interface Unit", consistent with the OSI model, to support differing topologies.
- o Gateway Development GSFC: Development and prototyping of a gateway to couple the networks being developed at JSC and GSFC.
- o Workstation Development LARC: Development of a state-of-the-art flight type workstation for evaluation in the Test Bed. Will incorporate findings from 'User Interface' tasks.
- o Space Qualified Computers JPL: An assessment of VHSIC technology and its impact/applicability to IOC. Also includes development of steps required to harden candidate processors against SEU and total dose affects.

o User Accommodation/Interface Definition - GSFC, JSC: GSFC to prototype actual 'tail circuit' designs suitable for interface with representative payloads and for use in conjunction with DMS Test Bed. GSFC to define access protocols/techniques for user interface to DMS services. GSFC to develop a set of candidate techniques for payload command and control scenarios to be evaluated on the DMS Test Bed. GSFC, JSC to define requirements for standard user interface capability for on-board and ground access to DMS services.

o DMS Simulator - MSFC: Development of a rudimentary DMS Simulator to be used by the ACS Test Bed.

o End-to-end Activity - JSC: The merging of the DMS Test Bed in an end-to-end sense with customer (GSFC), C & T, and JSC Ground Support Operations Center Test Beds. Planning and conduct of specific end-to-end testing using prototype ground workstations to establish ground flight controller interface requirements is also included.

o Color Flat Panel Display LaRC: In conjunction with the Navy, and Army (ERADCOM), NASA is funding/managing a 2-phase contract with Planar Systems to develop a full color, electro-eluminescent display panel. Phase 1 is the development of thin film primary phosphor technique; the phase 2 goal is the development of a 6" diagonal, 240 X 320 pixel, full color display.

5.3.1 Space Station Data Management Advanced Development Project Plan

The "Space Station Data Management System Advanced Development Project Plan" consists of 42 separable tasks assigned to various NASA Centers to provide a coherent program for the development of tools and technologies targeted to the On-Board Data Management and the Software Support Environment. A complete summary of the project plan tasks is provided in Appendix D.

A number of the tasks are dedicated to the development/implementation of an operational Data Management System Test Bed. The remaining tasks are summarized by general subject below:

- o Network Evaluation Ames Research Center (ARC): An ongoing capability to comprehensively evaluate the Goddard Space Flight Center (GSFC) Fiber Optics Data System (FODS) network (Task 7); this capability is being expanded to model the Johnson Space Center (JSC) Fiber Distributed Data Interface (FDDI) Token Ring system (Task 13) and the IBM unique network in development (IR&D).
- o Network Operating System JSC, GSFC: Both centers are to develop and prototype complementary NOS segments, derived from commercial products, compatible with the Open Systems Interconnections (OSI) Model. The two segments are to be combined into one NOS for use on either the GSFC or JSC network topologies.
- o International Standards Organization (ISO/OSI) Model Implementations: Commercial implementations and evolving industry standards are being comprehensively evaluated.
- o LISP Processor/Expert Systems JSC: Implementation of a LISP processor capability to host contractor developed "expert systems".

5.3.2 Survey Of Commerce Business Daily (CBD)

Two entries in the CBO were identified as independent from and supplemental to the activities of Section 5.3.1. These entry/activities are:

- 1. Non-Volatile Solid State Memory LaRC (CBD Pub. Date 9/7/84):
 Demonstration of a high data rate, non-volatile solid state memory. The scope of work encompasses definition of two functional models using Metal Nitride Oxide Semiconductor (MNOS) and Silicon Nitride oxide Semiconductor (SNOS) devices.
- 2. Optical Disk Recorder (10 Gigabit) GSFC (CBD Pub. Date 1/17/85): Development of ruggedized versions of commercially available technology. The initial phase will utilize available "write-once" devices then will address eraseable disk technology.

5.4 SSDS A/A Study Technology Development Candidates

In the course of the SSDS A/A Study, options developed in support of the functional requirements were evaluated within associated Task 3 Trade Studies to determine the preferred approach/technologies. In these trade studies, projected availability/development risk was a primary evaluation parameter. Consequently, the Task 4 system definitions include some "fall-back" solutions in cases where technically superior technologies were discounted because of development risk. To support a more comprehensive evaluation of the NASA plans, the set of advanced development candidates provided here includes not only technologies used directly in the Study system definitions but also those that, with intensive development, would provide a significant payoff for the Space Station Program. The total candidate list therefore consists of the following:

o Fiber Optic Network: The ANSI X3T9.5, a 100Mbps, ring topology, token access format network was selected for the On-Board local area network (LAN) design. This emerging standard must be prototyped/evaluated in support of preliminary SSDS design decisions.

o NIU: The Network Interface Unit for the On-Board system has been functionally defined to support OSI layer 1 through layer 4 services, and defined protocols, while satisfying the 100Mbps network data rate. This device must be prototyped and evaluated in terms of functionality and performance (throughput) in support of preliminary design decisions. Comparable devices being developed on DoD programs have demonstrated the NIU device to be in the critical path of the data processing system development. Hardware complexity is a design concern. This is reflected in terms of overall high parts count, high speed requirements related to the 100 Mbps fiber optic interface, a necessity for extensive built-in fault detection to maintain bus topology fault tolerance, and a layered design to accommodate technology insertion for post IOC.

o SDP: The standard data processor for the On-Board system was defined with a performance of 2MIPS and a 4MByte memory. Fault tolerance will be a requirement. VHSIC technology is anticipated with its performance and potential total dose radiation advantages. However the probability of a VHSIC SDP design to support IOC is questionable. Single event upset phenomenon will also remain an issue.

It is currently anticipated that the selection of an instruction set architecture must be made essentially at Space Station phase C/D contract start date. Since the onboard Data Management Subsystem must lend the other subsystem development, an off-the-shelf SDP might be a necessity. This implies that flight qualifiable a 2 MIP-4Mbyte design must exist within the next two years. If this is the case, then NASA must initiate near term activity related to the SDP.

- o Protocols/Formats: Of the ISO/OSI for the LAN's and the CCSDS definitions for the Space-Space, and Space-Ground communication links is a essentially a Program given. End-to-end design, in terms of coherent packet formats, packet switching and data distribution, will result in additional evolution of at least the CCSDS formats/structures. However, in order to meet space station requirements some changes to the detailed CCSDS formats must be made. Overt action is requested on NASA's part to cause this to occur.
- o Distributed Network Operating System: For the On-Board System, a networking, distributed processing architecture was defined supported by a distributed operating system. The DOS was functionally defined at a relatively high level and represents a significant development risk. A DOS of the space station complexity has never been designed and built before. Several systems in DoD programs have developed requirements and initiated breadboard level testing but none have been successful to-date. Partitioning of DOS functions between the SDP and the NIU along with what level of dynamic reconfiguration which can be allowed at the network level pose considerable alternatives. Network level redundancy management schemes may or may not be integral to the DOS design. Early function allocation and designs should be prototyped and exercised in a test bed environment.
- o Distributed Data Base Management System (DBMS): The extensive amount of On-Board and On-Ground data must be updated and manipulated with a defined data consistency and minimal query response times. A distributed DBMS was selected for the On-Board design but represents a development risk. Commercial developments of distributed DBMS's have been under development for several years but no significant capability releases are in use to date. The ability to maintain a real time distributed data base and concurrently to provide application software a reasonable quantity of historical snapshots of the same data as well as other correlatable data poses a challenging DBMS design.
- o Command/Resource Management: Allocation of resources to full payload complements and insuring that payload commands do not jeopardize the Station/Platform health and welfare or data quality of other payloads all while supporting telescience concept, is a major development item. The MDAC study made some in-roads on the subject but the concepts and methodologies

were not completed. This subject, in conjunction with mission scheduling is pivotal to the customer utility. To complicate matters further the considerations of privacy and security provide other constraints in the command management implementation. The dynamics of the onboard core and payload state makes for a tremendous bookkeeping problem along with convenient methods for predictability of resource utilization.

Both commands and resource definition require large amounts of data. However, the most difficult thing to cope with is the development of the algorithms to use this data in a predictive and deterministic manner. Prototyping of these algorithms should be initiated as soon as possible in order to develop implementation options. Telescience transparency of commands and actual command delays need to be determined in order to assess computing requirements in terms of throughput, local memory and mass memory requirements. The questions of what level of parallelism can be achieved because of a distributed processor architecture must be analyzed and options developed. This also could have distributed operating system requirements implications.

o Read/Write Optical Disk: Re-writeable high speed, random access, high volume, mass storage is a fundamental requirement for the On-Orbit elements. The optical disk technology is being strongly supported based on government/industry needs however there remains some IOC readiness risk. Key application areas benefiting from this capability are the following:

- A. Buffering of delayable payload data both in space
- B. On-board space station data base.

Buffering requirements for payload data ranges from $2x10^{11}$ bits for the space base to $5x10^{11}$ bits for the POP. Typical on-board space station data base mass storage includes manuals and procedures, software and scheduling information with requirements has been estimated to be in the order of $2x10^9$ bits.

Eraseable optical disk is a relatively new technology that has the high capacity and transfer rate characteristics that make it a potential candidate for both On-Board and Ground data buffer/storage applications. With proper funding this device can be developed and ready for use in the 1992 timeframe.

o User Test and Control (Interface) Language: Development of a high level control language is key to the user friendliness, productivity, and telescience drivers. Experience with the Launch Procession System (LPS) goal language usage has shown too much tight coupling of the use of the language with the details of the LPS hardware architecture. The MDAC team has had similar experiences with supposed generic commercial test systems. The ability to develop a space station test and control language, its attendant preprocessors and architecture dependant execution restrictions and further without requiring the user to emulate computer programmer thought processes is a real challenge. To achieve the desire implementation goals will be a development process requiring user participation and feedback. Rapid prototyping of many of the desirable language features should be initiated immediately in order for the evolving requirements to be developed. The design of the language has to accommodate a large spectrum of users encompassing software developers, test engineers, scientists and ground controllers. While this may suggest many subsets, the ability to integrate these subsets into cost effective and practical size wise software implementation will require innovative designs.

o Flight Qualified AI Processors/Expert Systems: These elements were not selected for the MDAC System Definitions however they represent significant gains in meeting productivity, and autonomy drivers in the areas of scheduling, maintenance (and trouble-shooting), etc. Expert systems have emerged in the commercial world to a degree that several implementations are being use in a highly improved productivity applications. However, the implementations have been accomplished using small 6-10 engineer/programmer teams. Several applications for the space station promise significant productivity improvements if both the algorithms/inference engine designs can be developed. One such candidate is scheduling of space station resources, whether it be for power, thermal control or crew time. Of significance is the

size of the development team required for this design. It certainly is an order of magnitude greater than has been done to date. To implement such a system onboard the space station would very likely require a flight qualified special purpose AI processor. While it is not expected that such processor will be available for IOC it is anticipated that they would be available shortly after IOC. Consequently, the significant increase in productivity and hence cost reduction is attractive enough for NASA to consider funding some development in this area.

o Color Flat Panel Displays: This technology, targetted primarily for On-Board workstations/displays, provides a distinct advantage over monochromatic displays which would result in gains in user/operator productivity. Color provides the potential for greatly increasing information coding capability and flexibility, as well as for reducing visual search time on complex displays. Secondly, luminance requirements can be reduced without a reduction in visibility of displayed information because of the addition of chromatic contrast. Color also provides for a greater degree of user productivity because of the aesthetic preference for working with color displays.

This technology is at an immature level. The advantages of brightness, contrast ratio, and viewing angle of thin film electro-illuminescent (TFEL) technology over the more mature color LCD technology dictate that advanced development effort be devoted to color TFEL. TFEL phosphors for red, blue, and green colors have been successfully developed, but have yet to reach performance levels obtained by manganese-doped zinc sulphide, the material used to produce yellow panels. Research effort needs to be directed toward development of new phosphors, and in particular, for the development of an efficient blue phosphor.

5.5 Conclusions and Recommendations

Evaluation of the NASA advanced development (A/D) plans against the candidate list provided in Section 5.4 led to the conclusion that a number of technologies need additional special emphasis and that the NASA A/D plans should be so adjusted. These technologies, with the A/D adjustment recommendations, are identified below.

- 1. Distributed Data Base Management This technology must be pursued by NASA to insure that some level of maturity is available for On-Board and Ground system utilization. It is recommended that an A/D Task be established to evaluate available commercial systems and develop a prototype system for evaluation on the JSC DMS Test Bed.
- 2. End-to-end protocols/formats Formats/protocols have a significant effect on efficiency, reliability and response times. It is recommended that Task 32 of the DMS A/D Project Plan be expanded to explicitly address evaluation of JPL/Contractor developed protocols/formats.
- 3. Command/Resource Management The approach to this technology in Section 5.0 the Task 4 Report represents a preliminary attempt to meet the necessary payload resource allocation and the system safety requirements while satisfying the telescience concept requirements. A considerable amount of work/definition was provided on the subject however the concept must be developed further to not only satisfy customer needs but also the operator/core system needs. It is recommended that two efforts be initiated to develop this technology. First develop a simulation test bed to evaluate command management algorithms. Once an algorithm or set of algorithms have been developed, then use these algorithms in an end-to-end functional simulation of the commanding process. This functional simulation would then be exercised in a telescience mode to provide the human interface design performance. Programmed delays in this simulation would be used to modify algorithms and iterate until an acceptable end-to-end design or concept is achieved. A second effort would be to develop and evaluate time critical command/resource management design options on the (end-to-end) DMS test bed.
- 4. Flight Qualified AI Machines On-Board AI (Lisp) machines to host 'expert system' applications represent significant potential autonomy/productivity increases. It is recommended that an A/D Task be established to sponsor this effort.

APPENDIX A CODE S/CODE T ALLOCATION

		<u>Implementation</u>	<u>Operation</u>
I.	ONBOARD SSDS		
	o Space Station	Code S	Code S
	o POP1	Code S	Code S
	o POP2	Code S	Code S
	o COP	Code S	Code S
II.	GROUND SSDS		
	o DHC	Code T	Code T
	o LZPF's	Code T	Code T
	o GSC	Code S	Code S
	o SSOCC	Code S	. Code S
	o SSEDC	Code S	Code S
	o P/F CC	Code T	Code T
	o P/F EDC	Code S	Code S

APPENDIX B NASA DMS TEST BED ADVANCED DEVELOPMENT PROJECT PLAN TASKS

NETWORK EVALUATION:

Task 1 - Extended Network Analysis - ARC

An ongoing network analysis capability that is being expanded to evaluate the performance of the GSFC FODS network being developed under Task 7. The ARC simulation is being expanded to model the FDDI Token Ring System under development by JSC (Task 13) as well as the IBM-unique network currently being developed under IR&D funding. It provides a NASA capability to conduct relevant, comparable analytical evaluations of candidate network configurations arising from Phase B. The ARC simulation will also provide the basis for a "calibrated" analysis tool which can be refined as Phase C/D progresses with the end result being a valuable verification capability to support post-IOC performance analyses as additional demands are made of the DMS Network.

DMS TEST BED:

Coordinated tasks to fund engineering and operation activities required to put the "test environment" in place and/or operate equipment supporting that environment.

Task 18 - DMS Test Bed Engineering - JSC

Task 19 - DMS Test Bed Operations - JSC

Task 27 - S/W Test Bed Operations - JSC

END-TO-END TEST BED:

Task 32 - End-to-End Test Bed JSC

Merges the DMS Test Bed in an end-to-end sense with customer (GSFC) and the Communication and Tracking, and JSC Ground Support Operations Center Test Beds. Also covers planning and conduct of specific end-to-end testing using prototype ground workstations to establish ground flight controller interface requirements.

DMS SIMULATION:

Task 38 - Data Subsystem Network Technology - MSFC

Provides funds for completion of a rudimentary DMS Simulator to be used by the Attitude Control System (ACS) Test Bed. The DMS/User ICD presently used for the DMS Test Bed is being provided MSFC for guidance in structuring a representative interface between the simulator and ACS.

NETWORK OPERATING SYSTEM:

Complementary segments from JSC and GSFC for the study and prototyping of Network Operating Systems derived from commercial products that accomplish the software functions of the ISO model. The intent is to segment those tasks such that each center "delivers" complementary elements of a standard NOS with the further requirement the two products combine, consistent with the appropriate ISO layering, to form one NOS for use on either the GSFC or JSC network topologies.

Task 8 - GSFC NOS (ISO Layers 3 & 4)

Task 29 - JSC NOS (ISO Layers 5, 6 & 7)

Task 23 - SSE System Software Studies - JSC

JSC to perform study/prototyping activity investigating the applicability of commercially available O/S to on-board O/S and NOS requirements.

NETWORK/BUS INTERFACE UNITS:

Task 7 - Optical BIU - GSFC

GSFC to design/prototype a hardware BIU required for FY-86 testing of the GSFC Star Bus system. That BIU is being effectively used as a "pathfinder" for specific study activities conducted under Tasks 8, 13, and 29.

Task 12 - MCNIU - JSC

JSC to produce a commercially-derived device, referred to as the Multi-compatible Network Interface Unit (MCNIU) which will provide potential Test Bed users with multiple options for transparent interfacing with different topologies and/or differing degrees of computational support consistent with accepted layering defined by the ISO/OSI model.

GATEWAY:

Task 9 - Star to Ring Gateway - GSFC

GSFC to develop and provide a prototype gateway to effectively couple the 'star' and 'ring' networks being developed at GSFC and JSC respectively. Will offer an opportunity for both centers to utilize the gateway in support of the DMS Test Bed end-to-end evaluations and to explore some specific issues known to exist relative to gateway implementation.

ISO/OSI MODEL IMPLEMENTATION:

Task 13 - DMS Interface Standards - JSC

JSC to perform the study and prototyping activity to define actual electronic implementations of the lower two layers of the MCNIU. Commercial implementations and evolving industry standards are being comprehensively evaluated for applicability to the required designs.

AI/EXPERT SYSTEMS:

Task 15 - DMS Expert Systems - JSC

Investigation of expert system technology for applicability to Automated Fault Detection and Isolation system.

Task 25 - Expert Systems - JSC

Develop and prototype an expert system of the type anticipated for use on Space Station.

SPACE QUALIFIED COMPUTERS:

Task 14 - Processor Technology - JSC

Essentially a VHSIC technology assessment with a final report advising NASA of the anticipated impact and applicability of VHSIC technology to IOC. This (FY-87 funded only) task takes advantage of the unique expertise of JPL in the area of Space Radiation effects on electronic devices. The intent is to have JPL provide a detailed definition of any steps required to harden a candidate processor against single event upset (SEU) and total dose effects.

Task 41 - Flight Processor Space Qualification Definition - JPL

Develop a program development plan for space qualifying a specific processor based on Level B analysis.

HOL/ADA ACTIVITIES:

Task 31 - Ada Target and Benchmark - JSC

Provides results such that the SSE designers are given a set of guidelines relative to the use of Ada for specific purposes. This task continues with increased focus on the very detailed implications of actually using Ada in real-time applications requiring rigorous verification techniques.

Task 39 - Ada Development Tools - MSFC

Procures development tools (cross-compilers, editors, etc.). Also supplements MSFC's capability to implement the ACS applications code in Ada.

Task 40 - Payload Simulator - NSTL

Payload simulator will be designed from information provided by GSFC, MSFC, and JSC, and will provide the end-to-end Test Bed with capability for conducting representative "closed loop" operations studies as well as assessing the interaction of various types of payloads on the onboard DMS elements. Task 40 is augmented with sufficient funds to allow NSTL to code the P/L Simulator using Ada.

Task 21 - Ada Evaluation - JPL

JPL support to JSC in evaluation of Ada as a Space Station High-Order Language (HOL). Task will develop a set of metrics and a decision matrix for the evaluation process.

Task 10 - Evaluation of Ada For Users - GSFC

The objective of the task is to acquire advanced software support environments related to Ada for evaluation. Ada application will be evaluated through the use of selected pilot applications for GSFC SS functions.

USER ACCOMMODATION/INTERFACE DEFINITION:

Task 3 - Customer Interface Adapter - GSFC

Prototypes actual "end-circuit" designs suitable for interfacing with representative payloads GSFC has identified. Will provide the DMS Test Bed with the capability early in phase C to actually accommodate selected real payload electrical interfaces for "high-fidelity" point design trade studies.

Task 4 - User Data Flow Management - GSFC

Produces and evaluates user access protocols/techniques for the (user transparent) management, storage and retrieval of payload, engineering, and ancillary data from distributed data bases.

Task 5 - User Interface Technology - GSFC

A coordinated effort with Tasks 6, 33, and 34 to support the SSE RFP with definition of requirements for a standard user interface capability suitable for onboard and ground access to DMS services. Encompasses the total spectrum of "users" from on-board system and payload control to ground telescience operations as well as ground pre-launch checkout operations.

Task 6 - P/L Cmmd and Control Techniques - GSFC

Produces a set of candidate techniques for P/L Command and Control scenarios that will be eventually evaluated in the end-to-end test bed environment.

Task 33 - User Interface Studies - JSC

Develops a common user interface to access On-Board/Ground SSDS segments. Integrates GSFC, KSC and JSC user interface language (UIL) tasks.

Task 34 - SS Operations Language - KSC

Specifies operations language and associated utilities in support of test and integration community.

WORKSTATIONS/DISPLAYS:

Task 16 - Displays & Controls - JSC

Provides a set of requirements for SS Displays and Controls and evaluates commercially available D & C devices.

Task 37 - Control/Display Interface Technology - LaRC

Provides a state-of-the-art flight-type workstation for evaluation in the Test Bed environment with the implicit acceptance that the device should evolve into an actual part of the Test Bed environment by supporting detailed development of operational scenarios during the early Phase C design phase. Its design will incorporate recommendations and lessons learned from Tasks 3, 4, 33, and 34.

SSE ACTIVITIES:

Task 22 - S/W Engineering Support - JSC

Provides direct analytical support required for synthesis of SSE requirements into coherent formulations suitable for end-item products.

Task 26 - S/W Test Bed (SSE Computer) - JSC

Procures a host mainframe computer for the Test Bed to provide a prototyping capability for the evaluation of SSE S/W.

MISCELLANEOUS:

Task 2 - Enhanced Telemetry - GSFC

This task will evaluate high rate Data Handling Center requirements. An SS requirements analysis with DHC options mapped to the requirements/assumptions plus a modular architecture design derivation will be performed as the first two sub-tasks. Expansion methodology and technology drivers will be identified to weigh the On-Board vs Ground data handling options, (packetization, calibrations, etc.).

Task 11 - Subsystem Test and Checkout - JSC

This task will investigate techniques and procedures for monitoring and evaluating the performance of spacecraft subsystems. Areas of investigation include automated fault detection and isolation, data base design, data management systems and crew interfaces.

Task 28 - SIB Development - JSC

Prototypes a Simulation Interface Buffer (SIB).

Task 35 - Software Reliability Technology - LaRC

This task will supplement on-going research to establish the reliability effectiveness of software fault-tolerance techniques.

Task 36 - Optical Transceiver Module - LaRC

Funds the development of a fiber optic transmitter/receiver to satisfy a common set of requirements for both NASA and DoD. The transceiver will cover the data range from 200 to 1000Mbps.